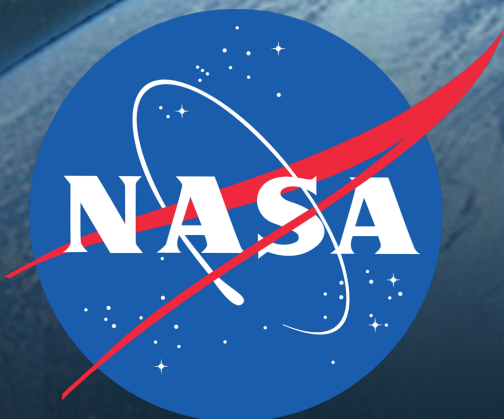




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Hemisphere/Wing Grid Generation Codes



Prepared by Hiroaki Nishikawa for
Solver Technology for Turbulent flows (STT)
04-08-2017.

Available codes



Hemisphere-cylinder grid generation code

`hcf_hc_v7p5.f90`

3D-wing grid generation code

`hcf_wing_v3p3.f90`

Topologically equivalent to HC grid.

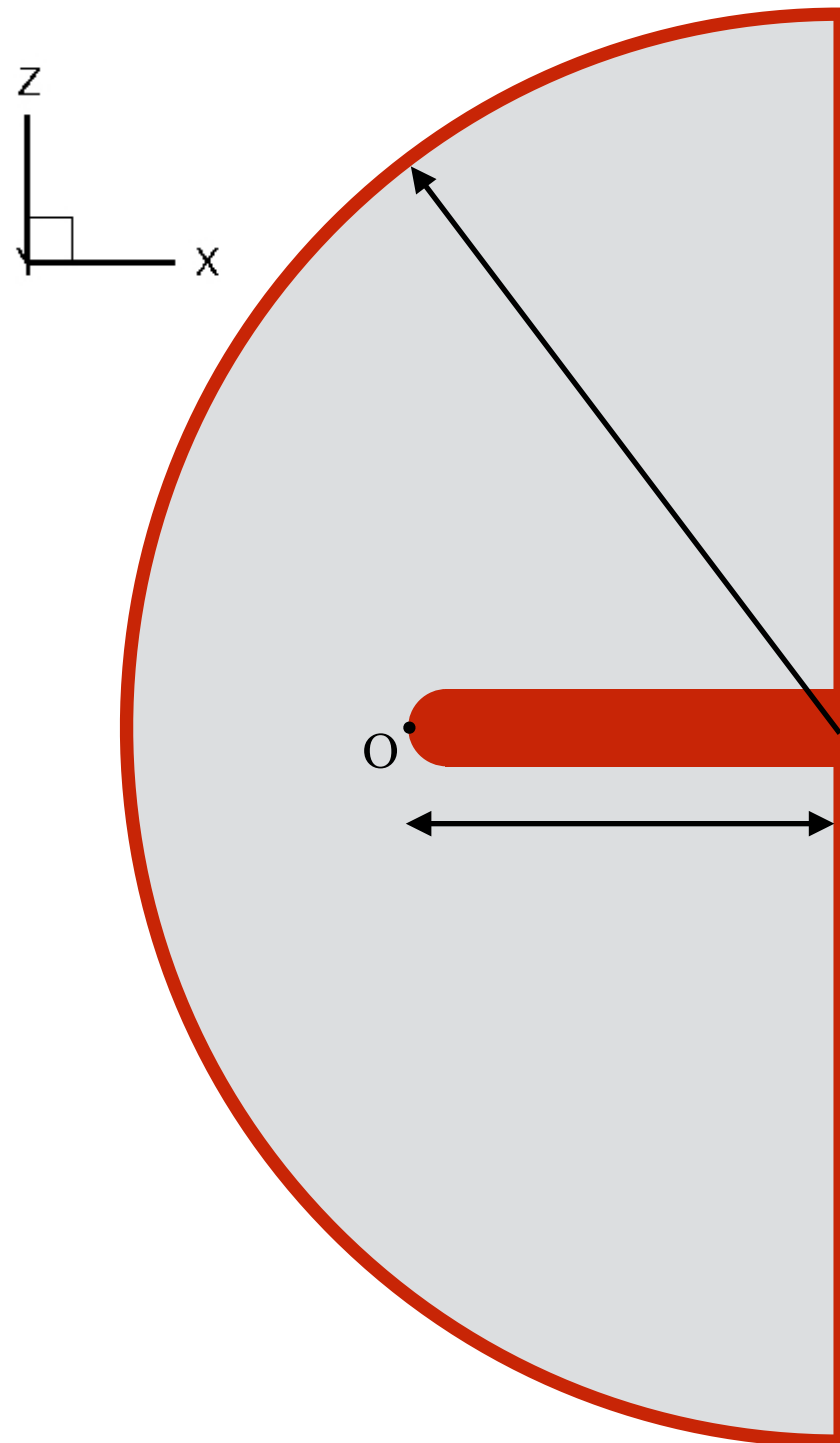
Regular coarsening code

`hcf_coarsening_v2p1.f90`

This program regularly coarsens HC or 3D-wing grids.

Hemisphere Cylinder

hcf_hc_v7p5.f90



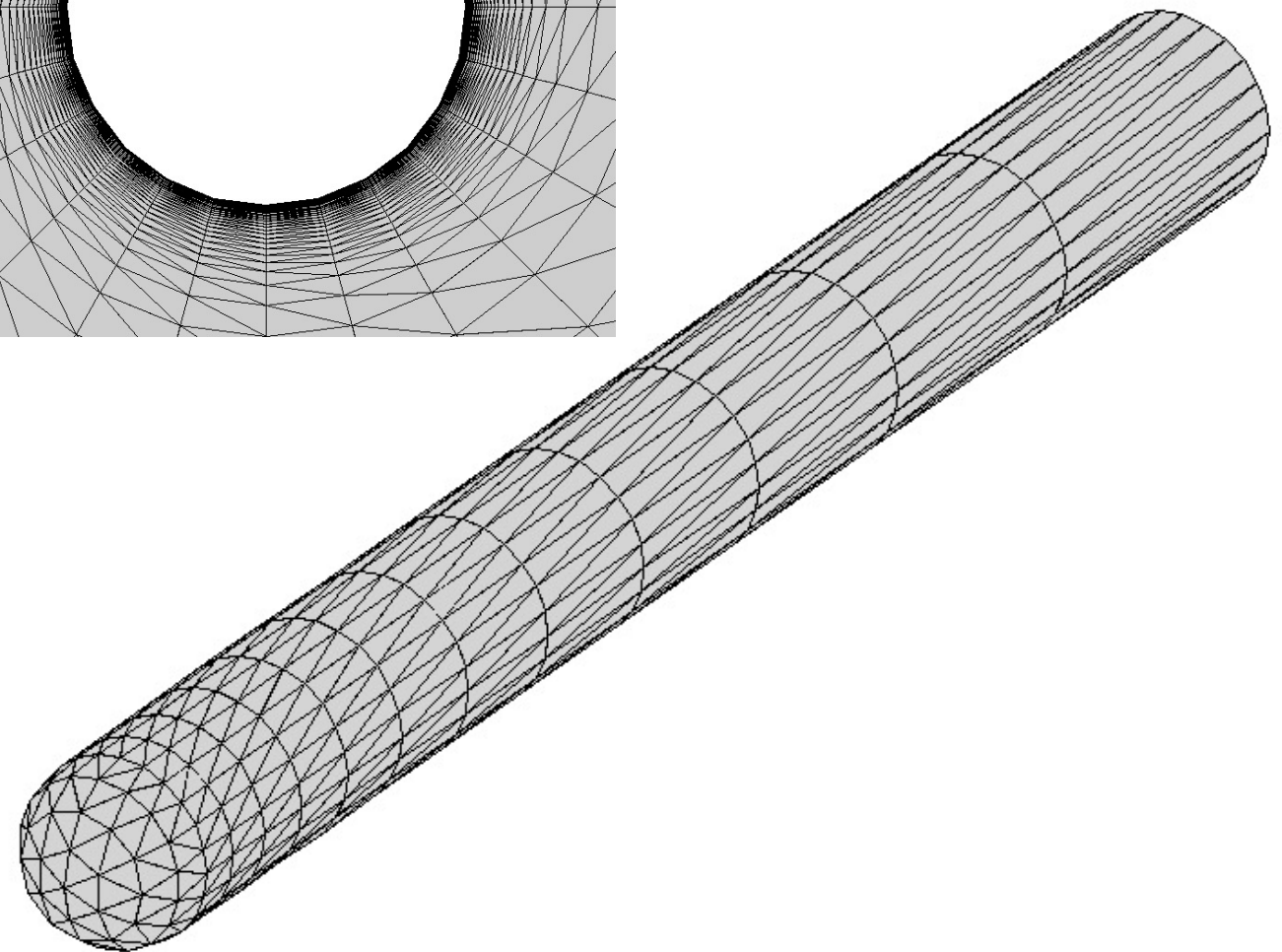
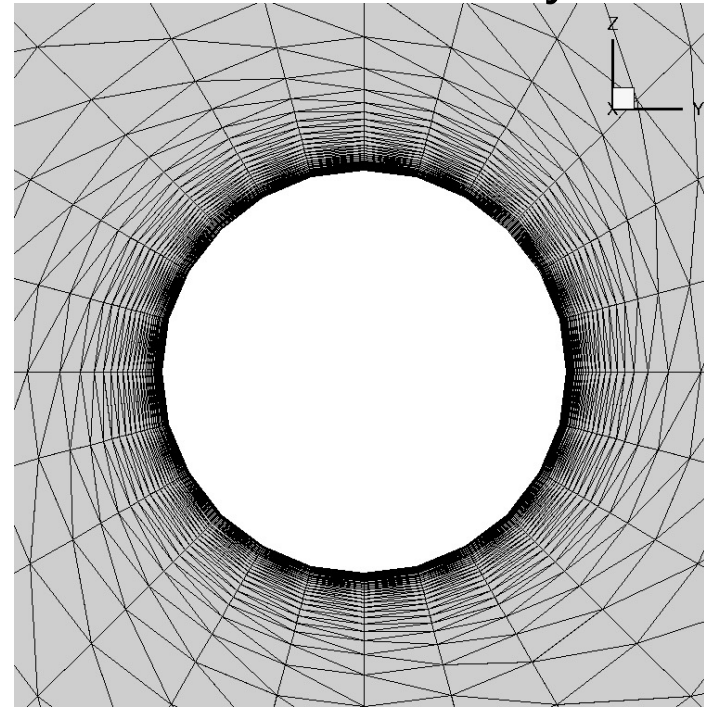
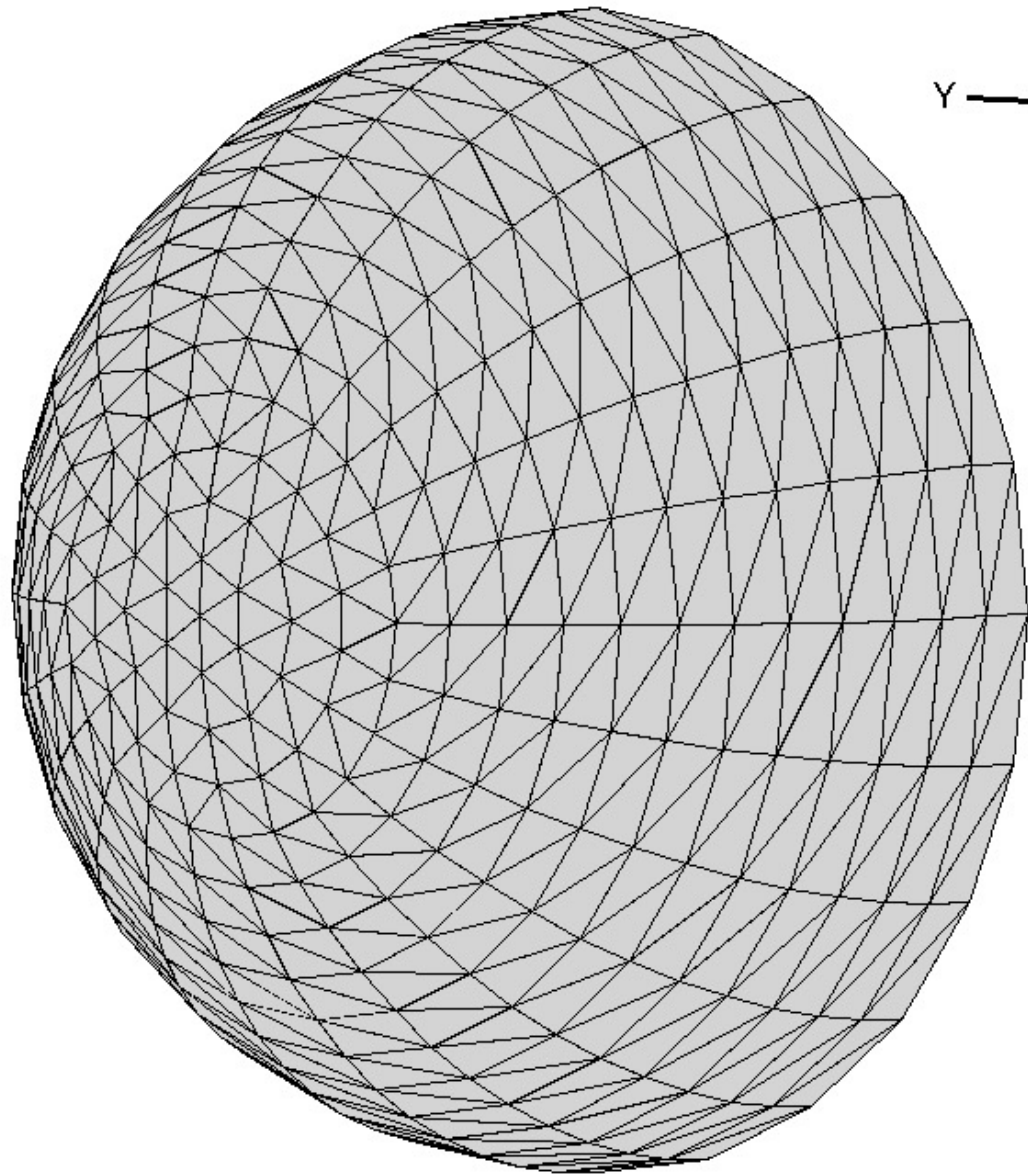
Hemisphere outer boundary:

Hemisphere-cylinder:

Apex at the origin

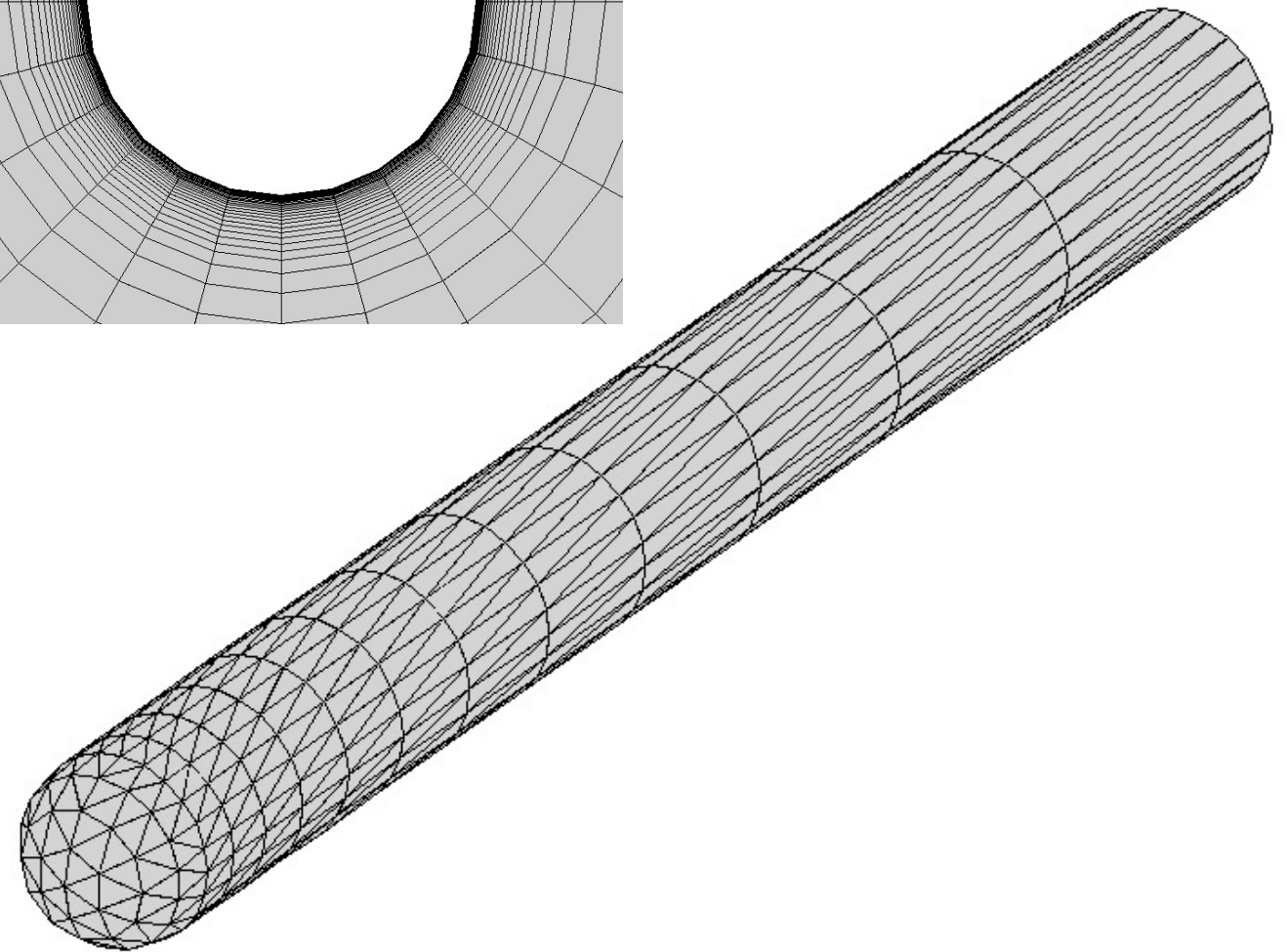
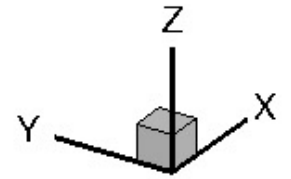
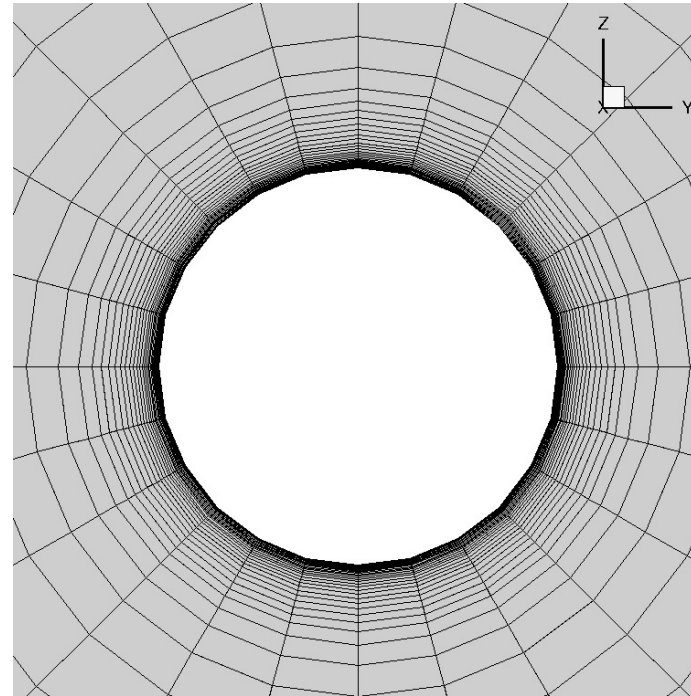
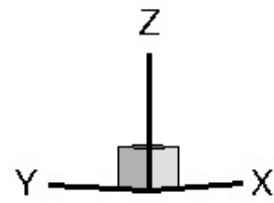
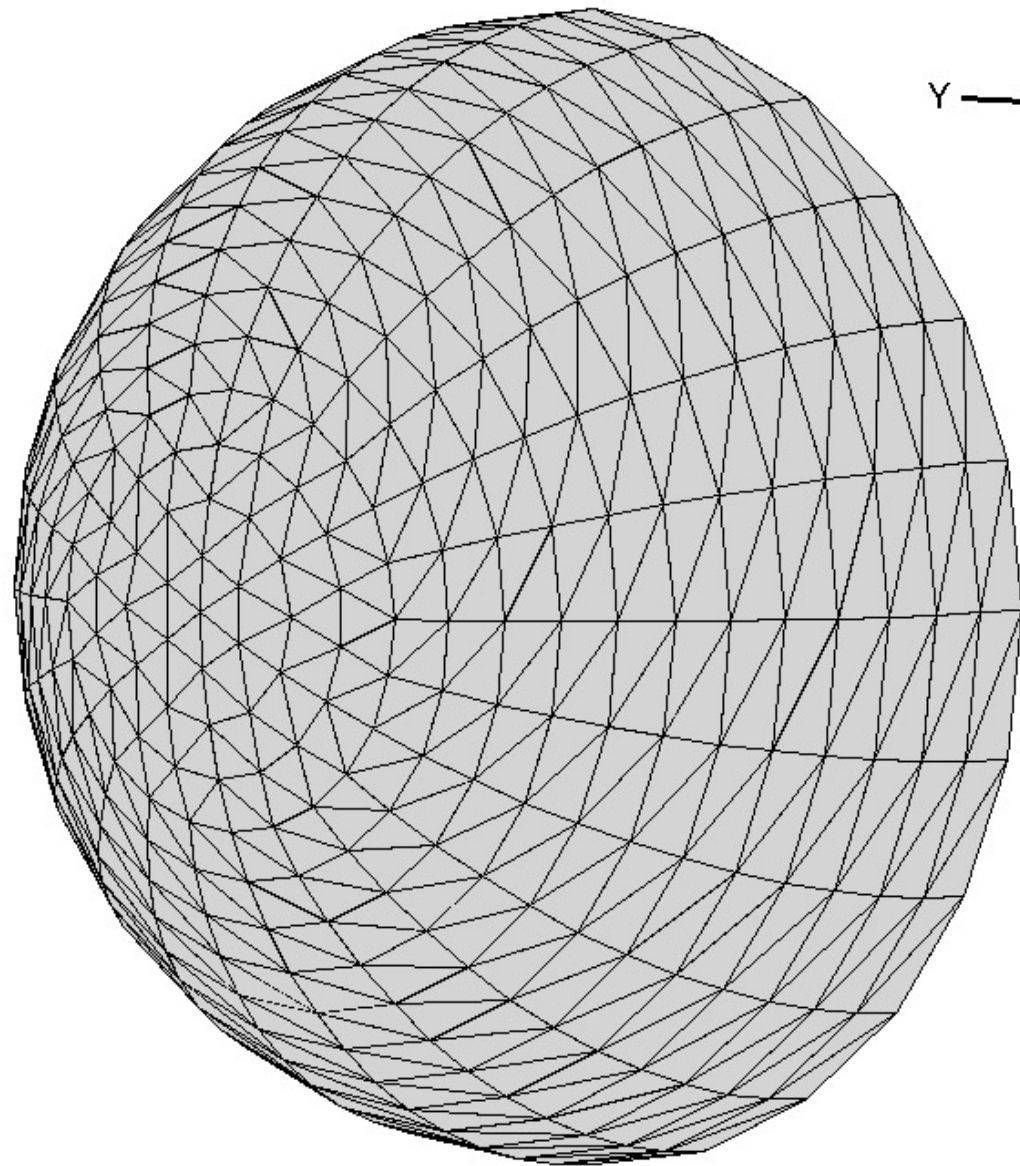
Radius of the hemisphere = 0.5

Outflow boundary



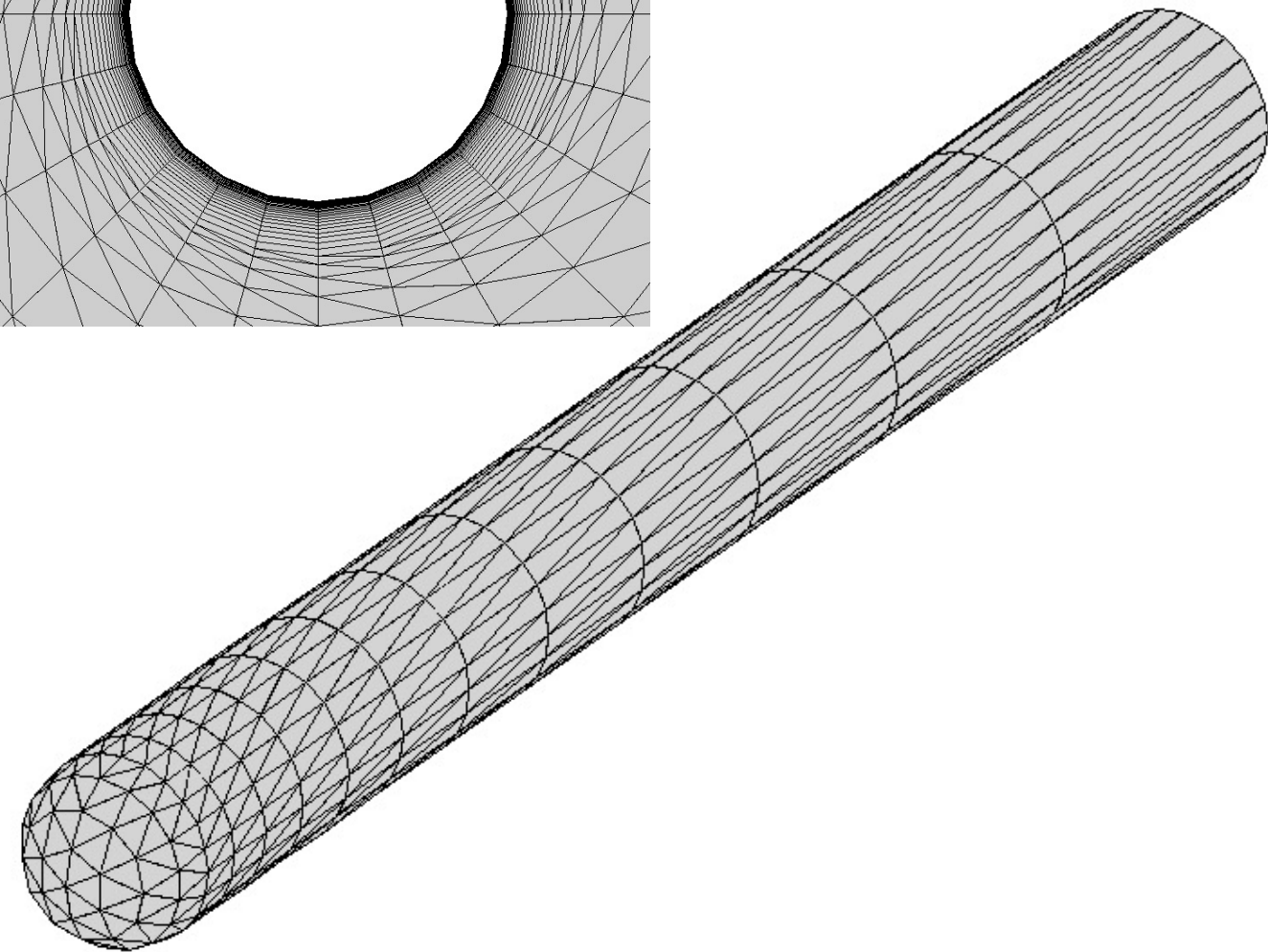
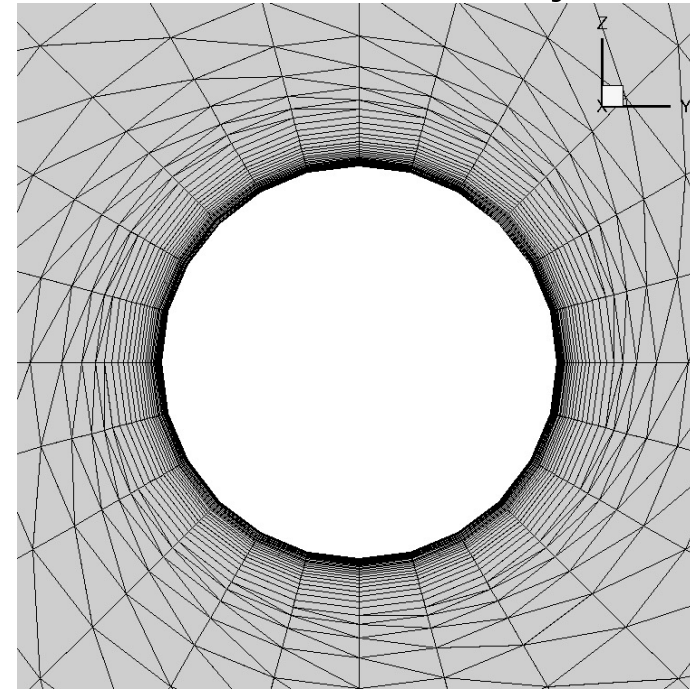
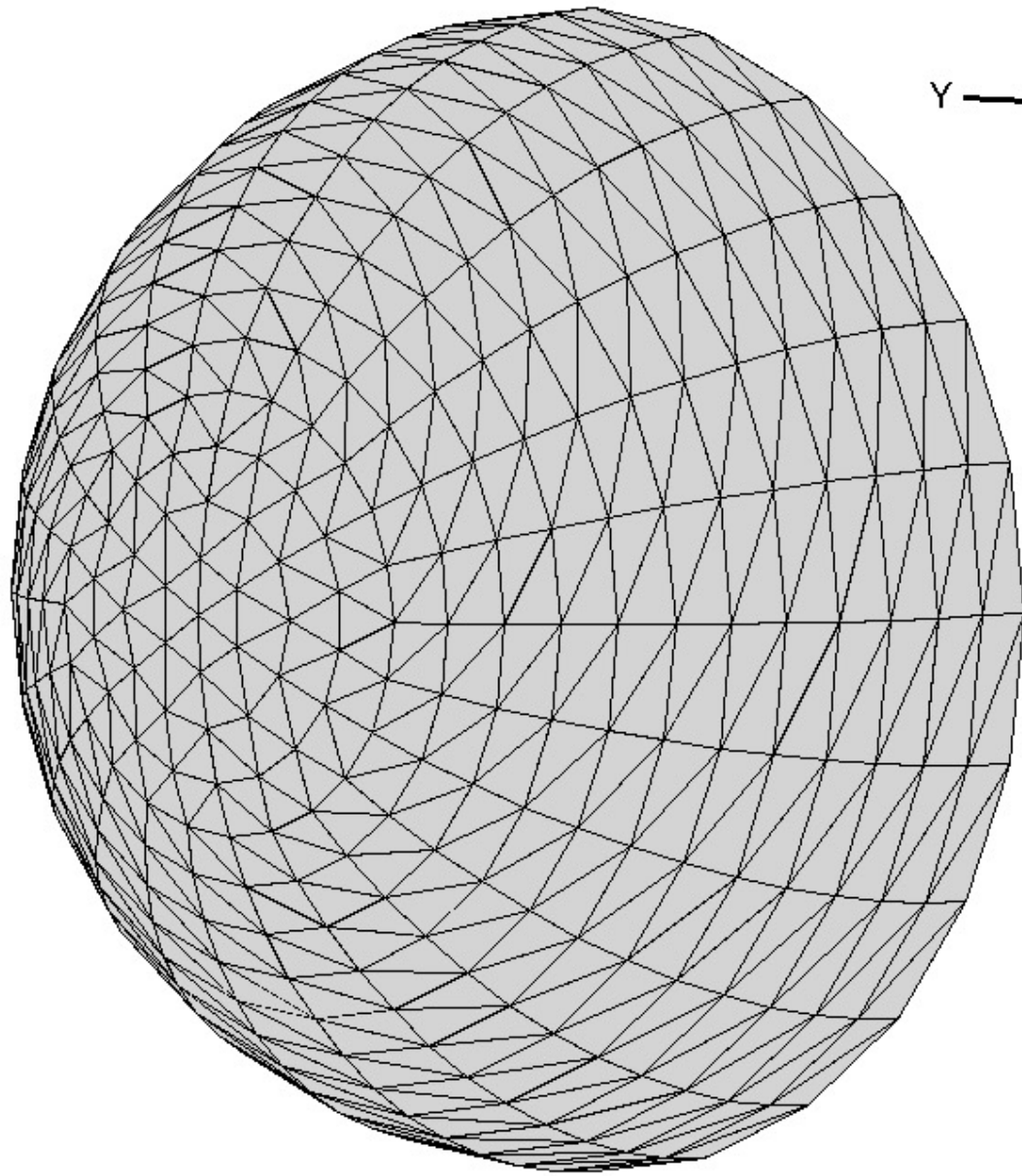
Prism

Outflow boundary



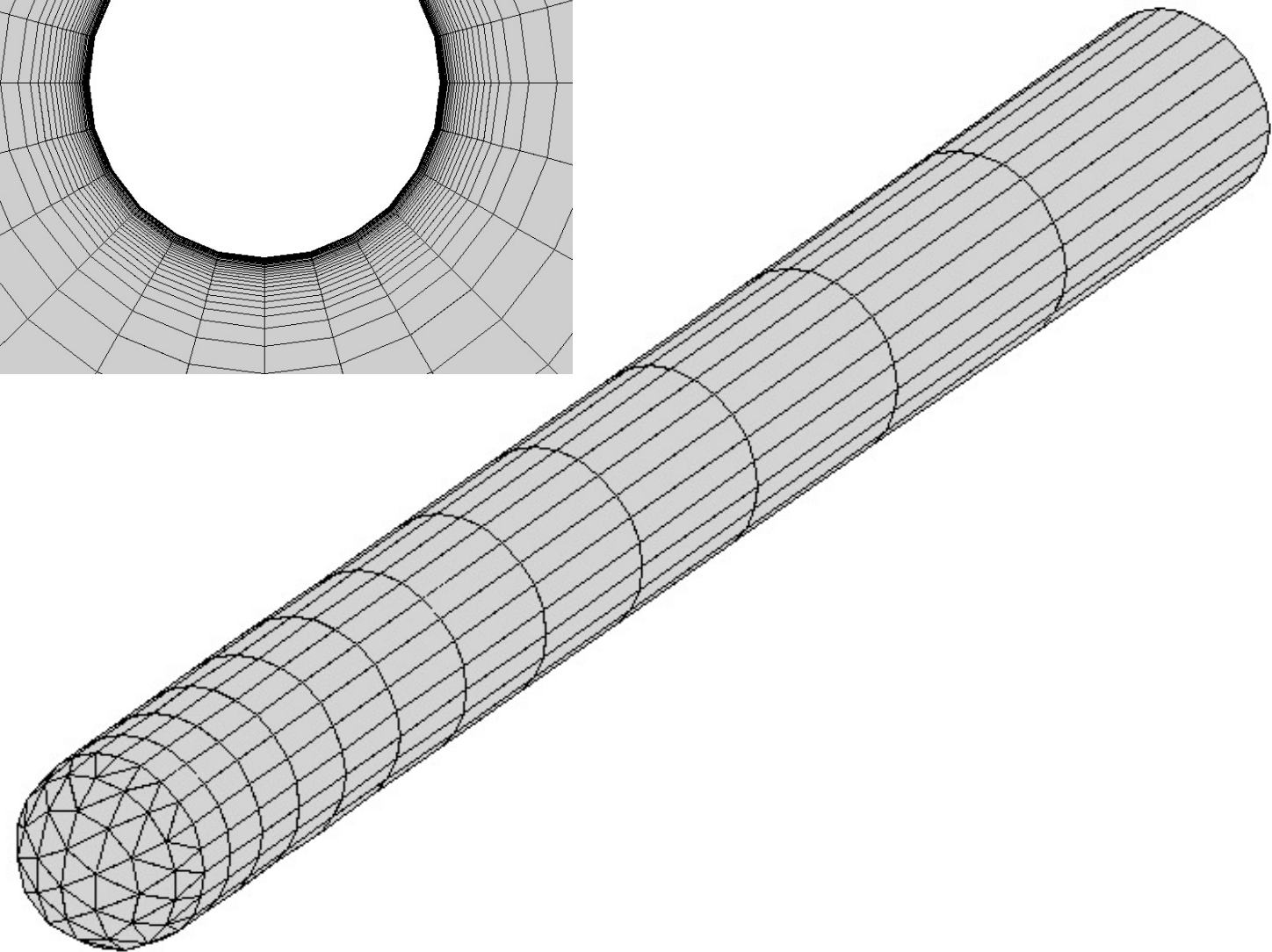
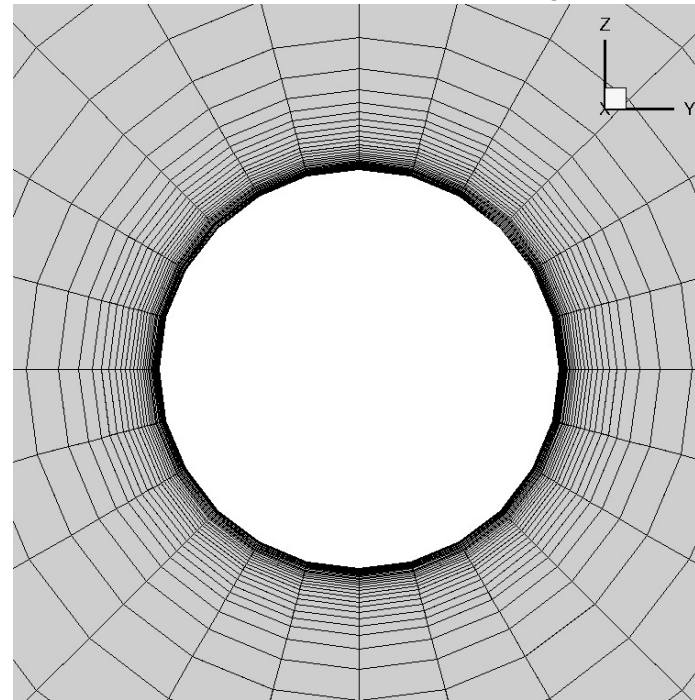
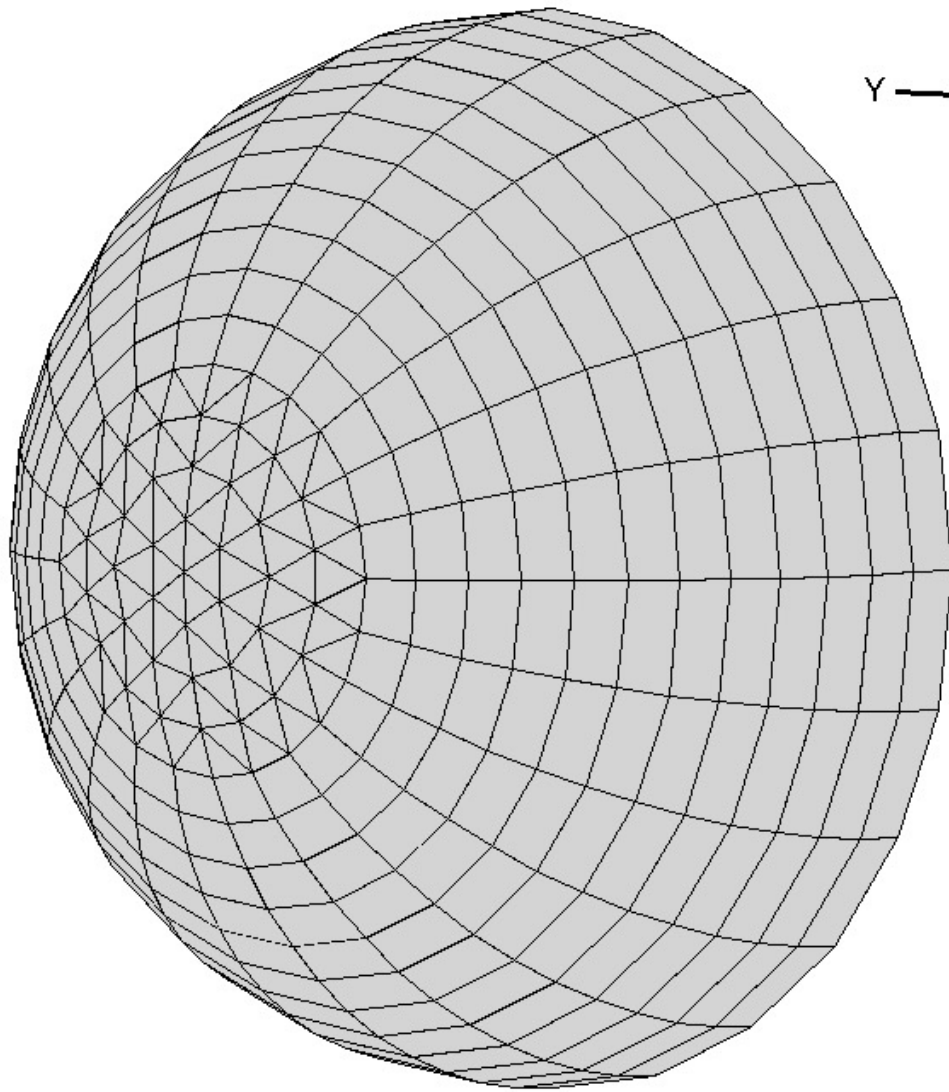
Prism-Tetra

Outflow boundary



Prism-Hex

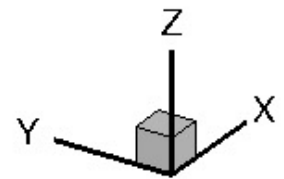
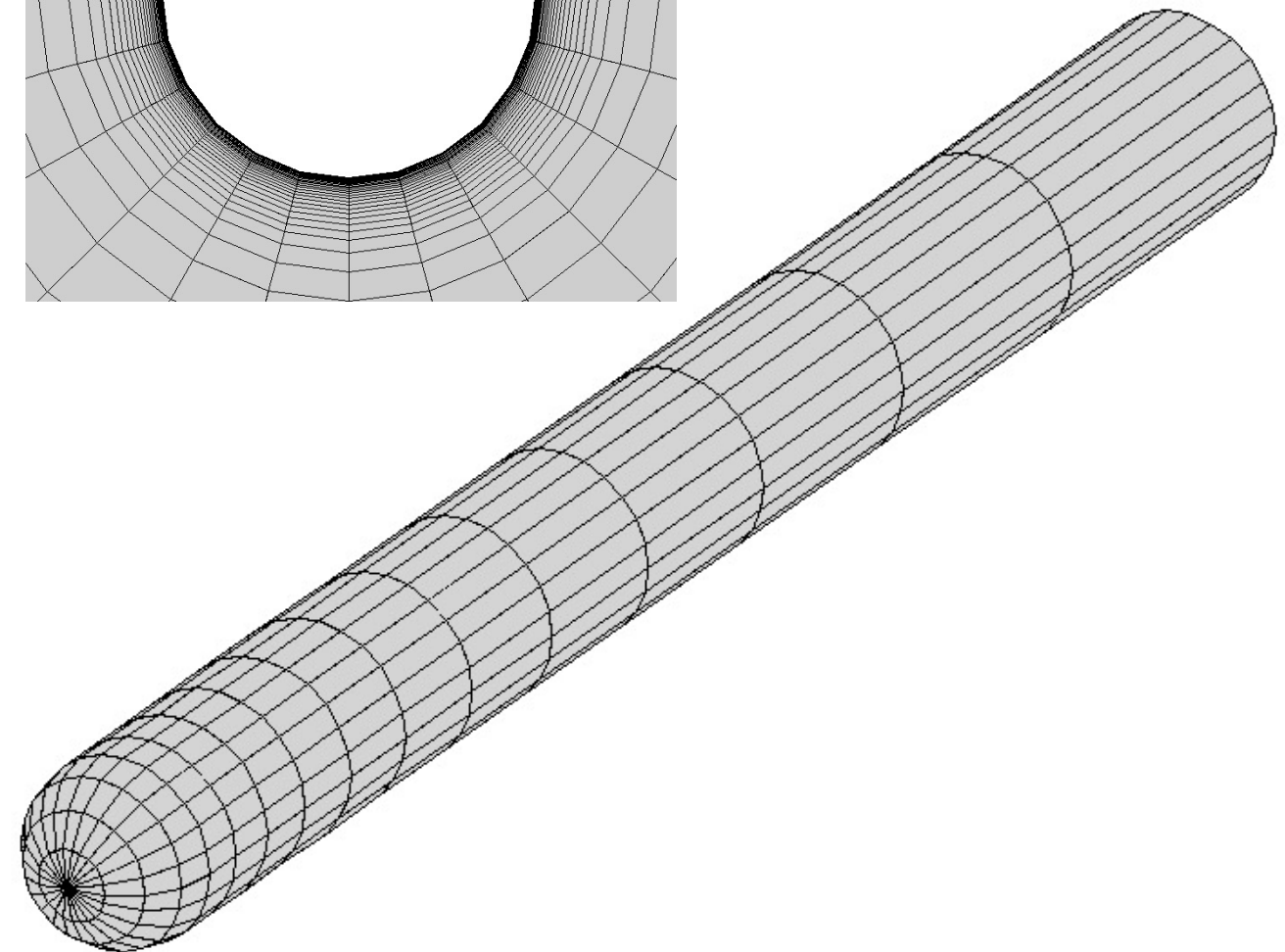
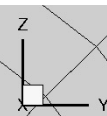
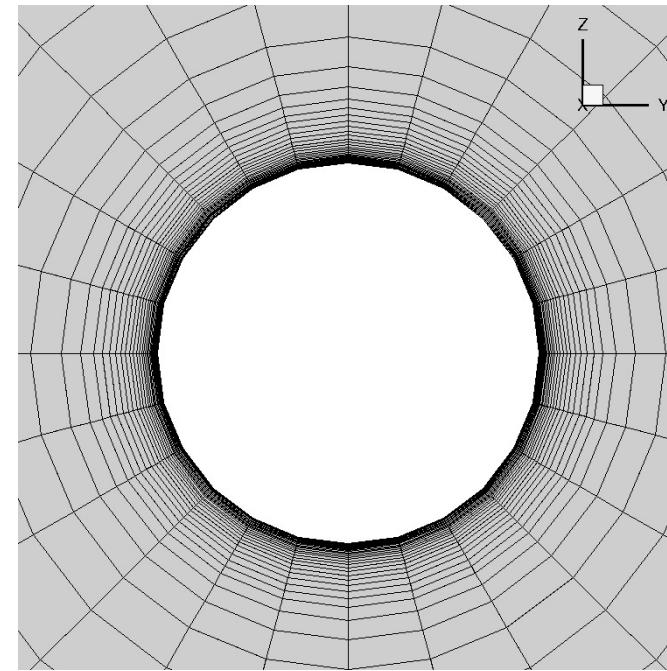
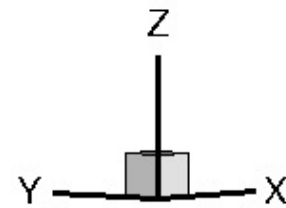
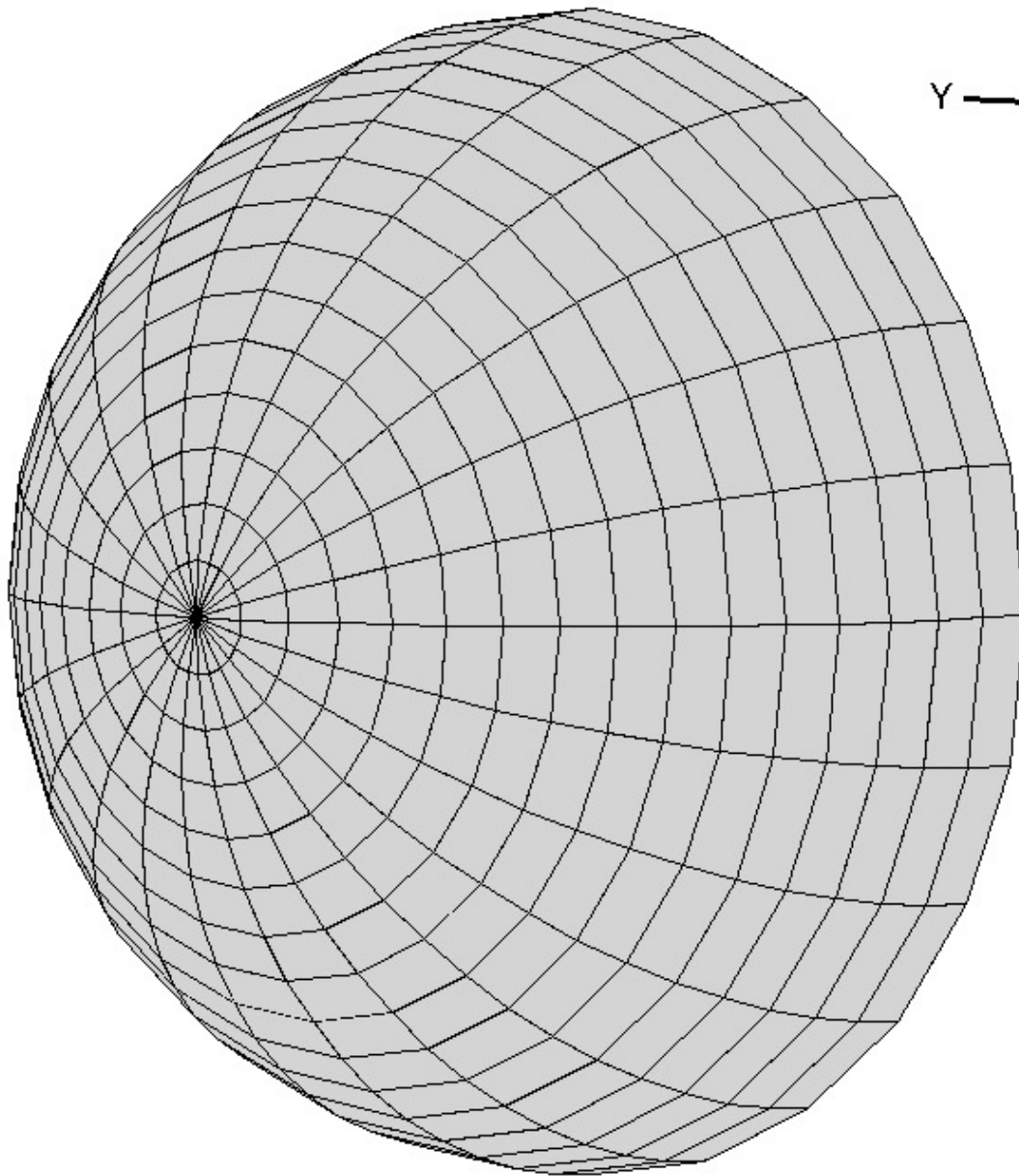
Outflow boundary



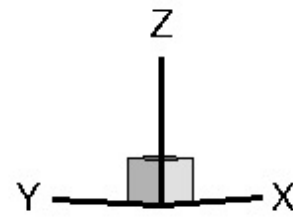
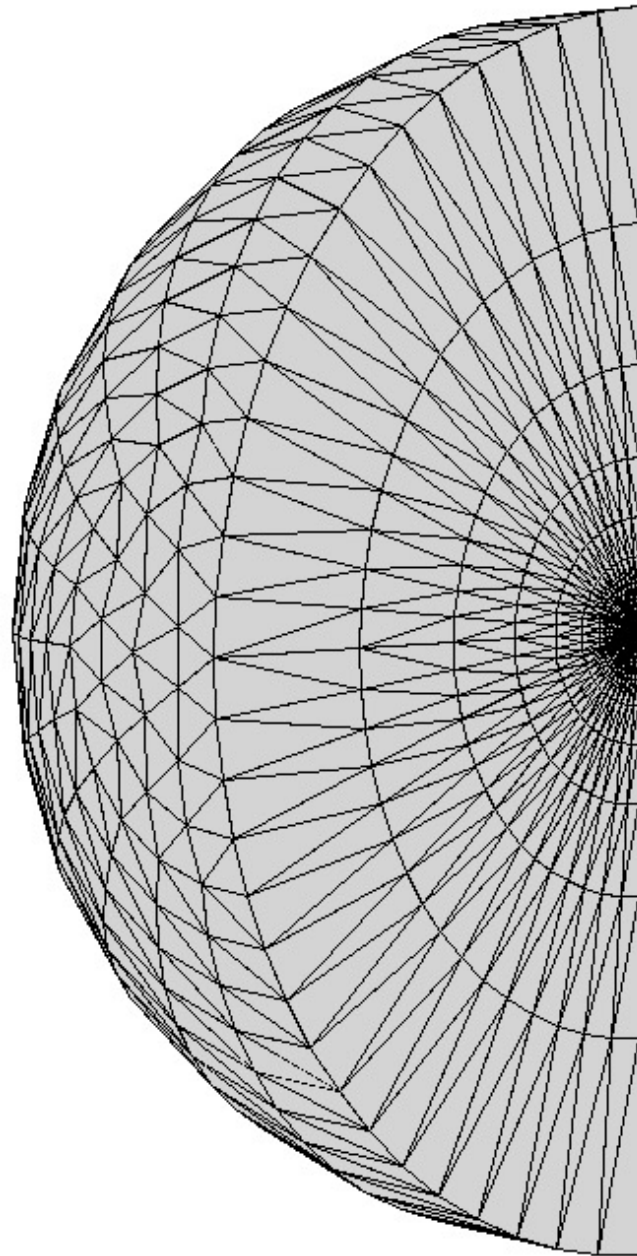
Structured (Hex+prism)



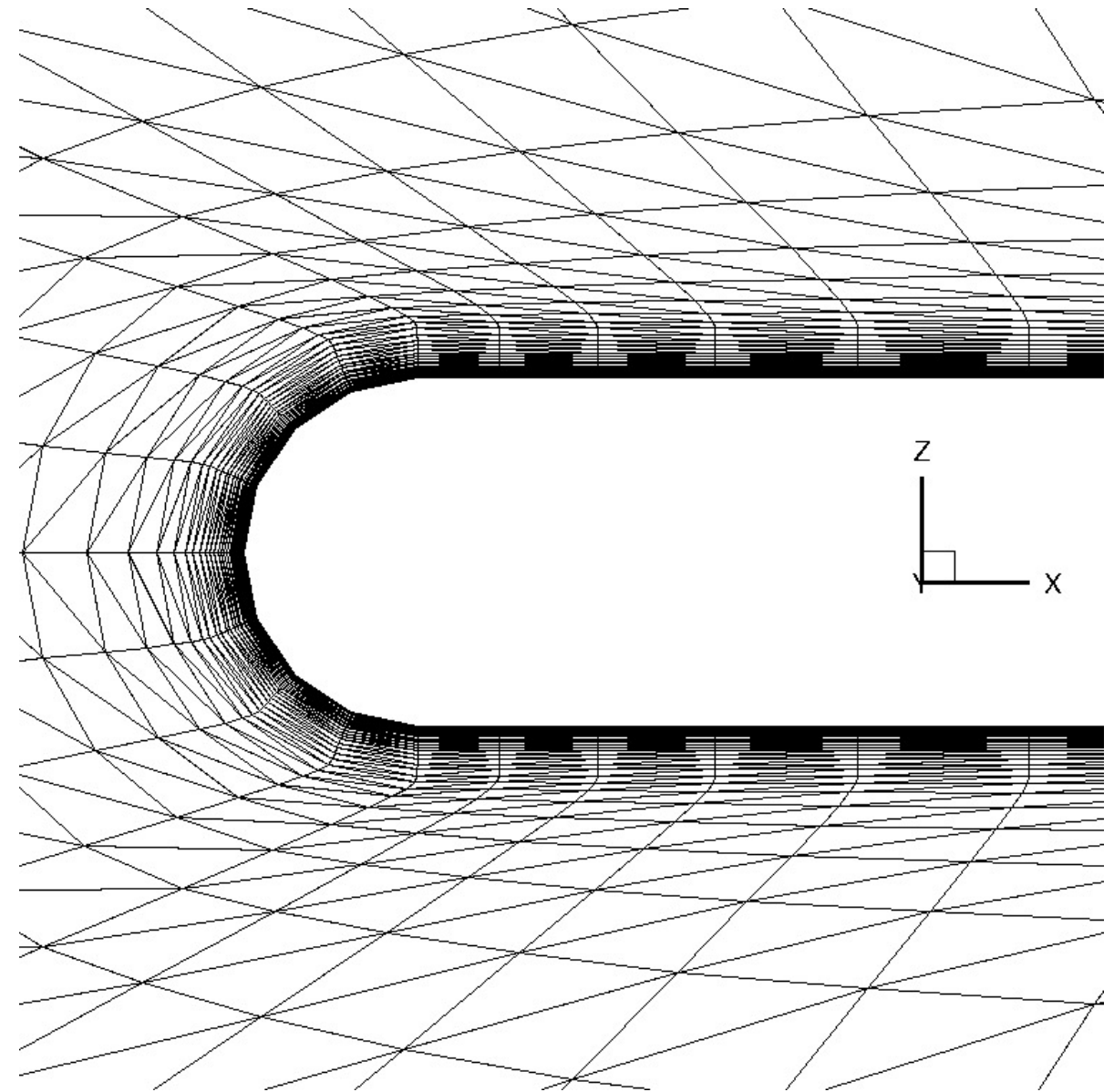
Outflow boundary



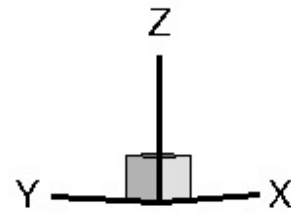
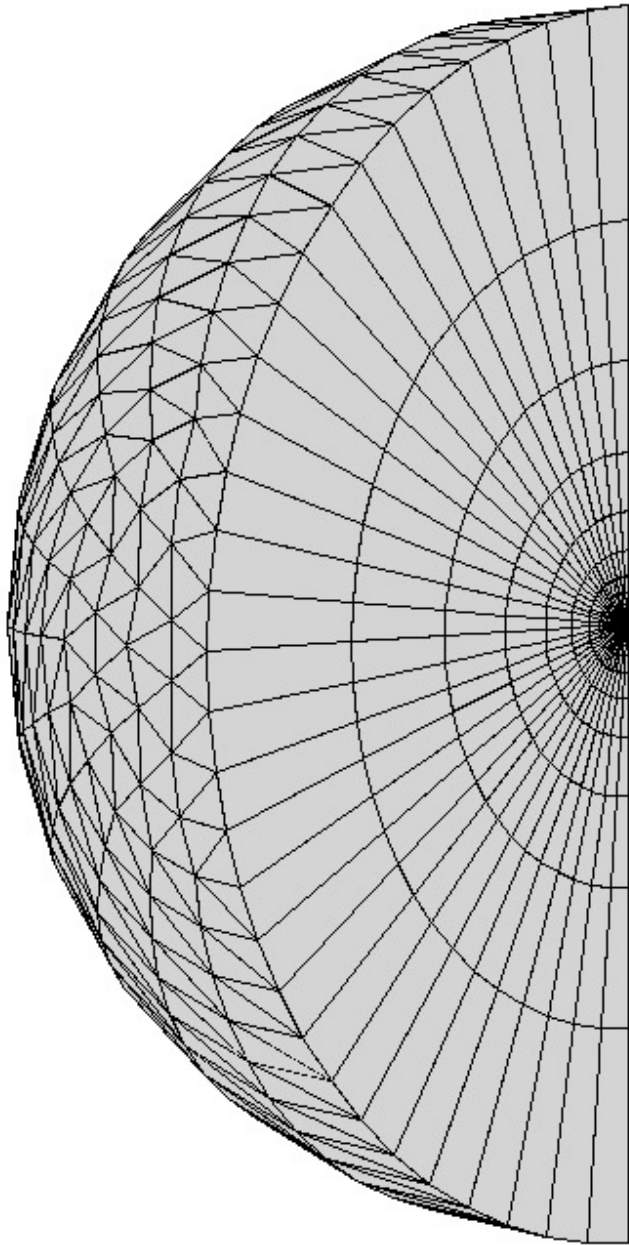
Half: Tetra



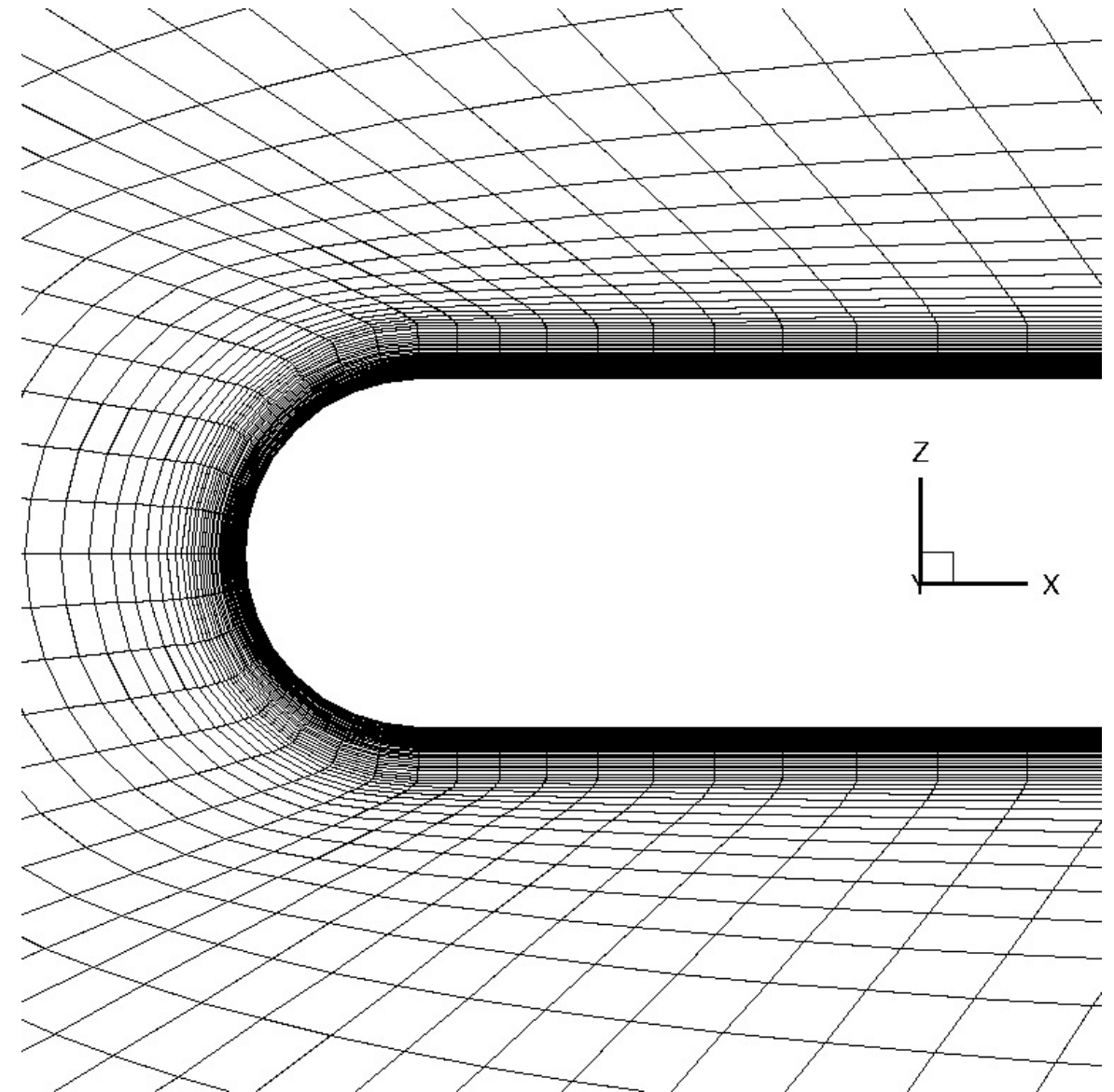
Symmetry plane



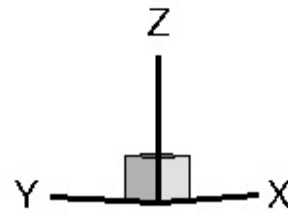
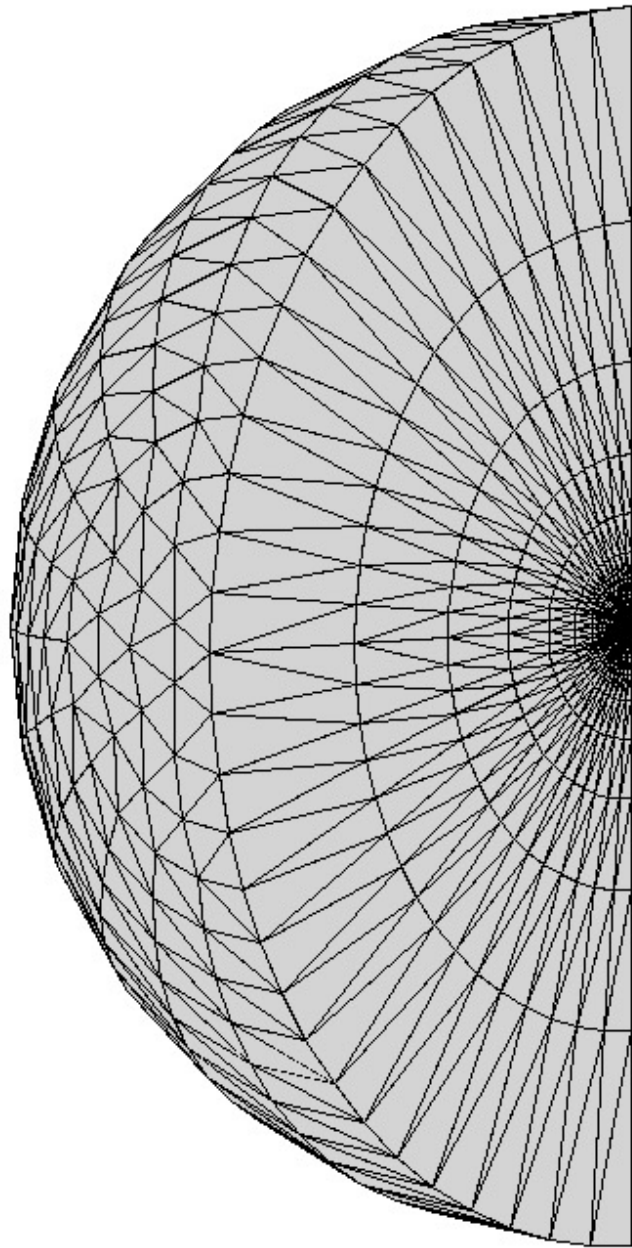
Half: Prism



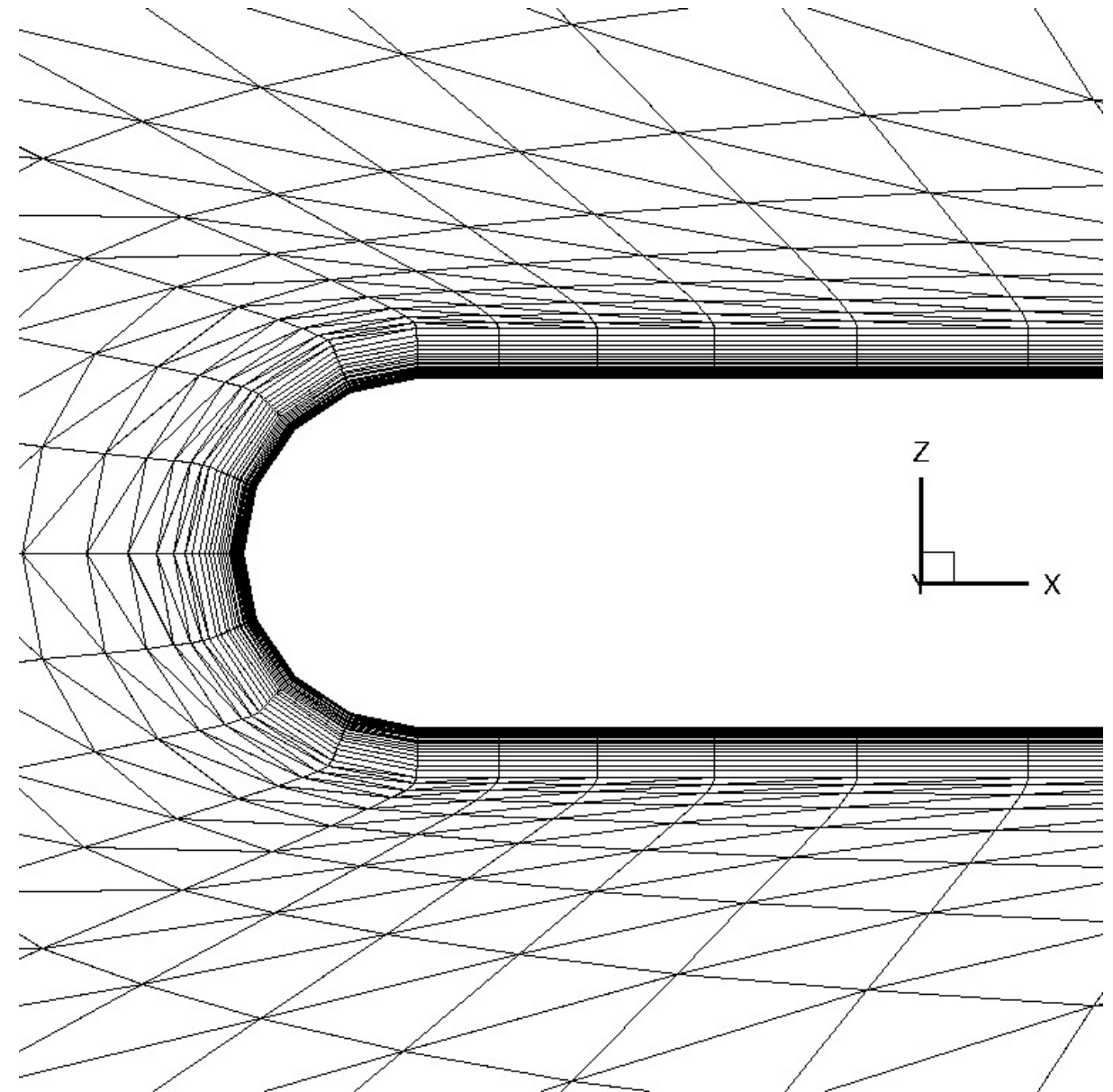
Symmetry plane



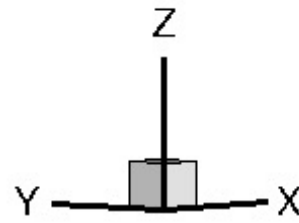
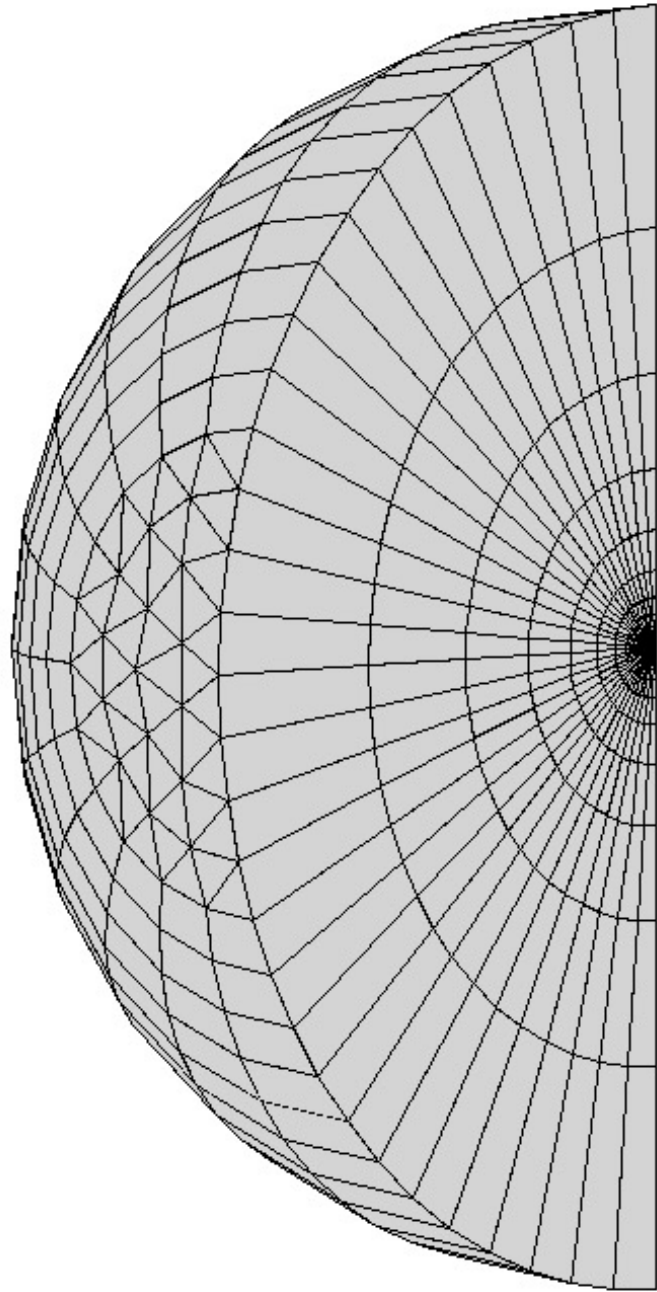
Half: Prism-Tetra



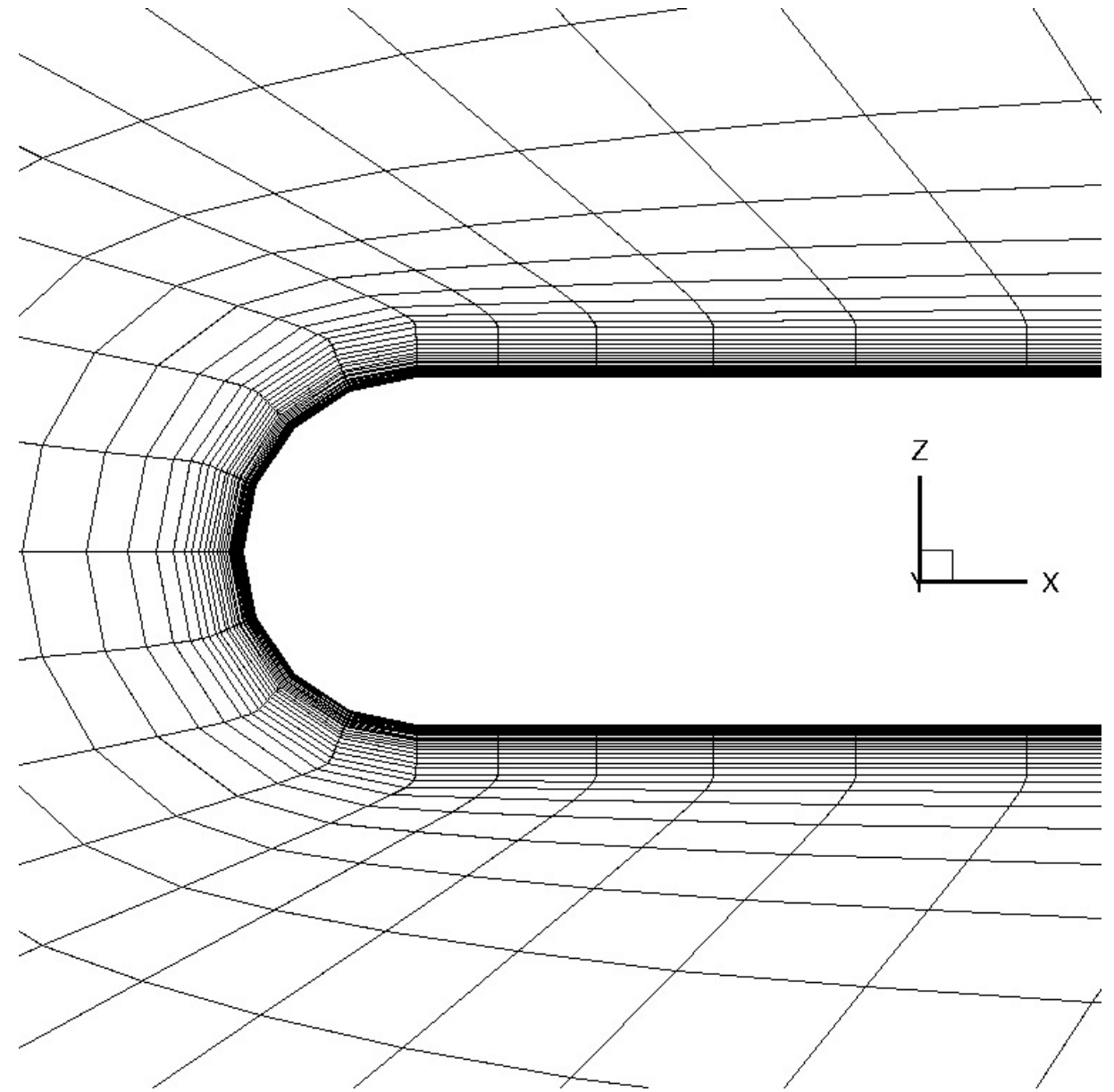
Symmetry plane



Half: Prism-Hex

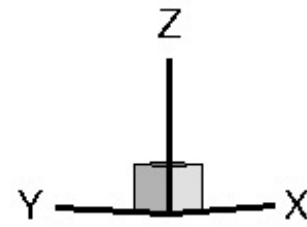
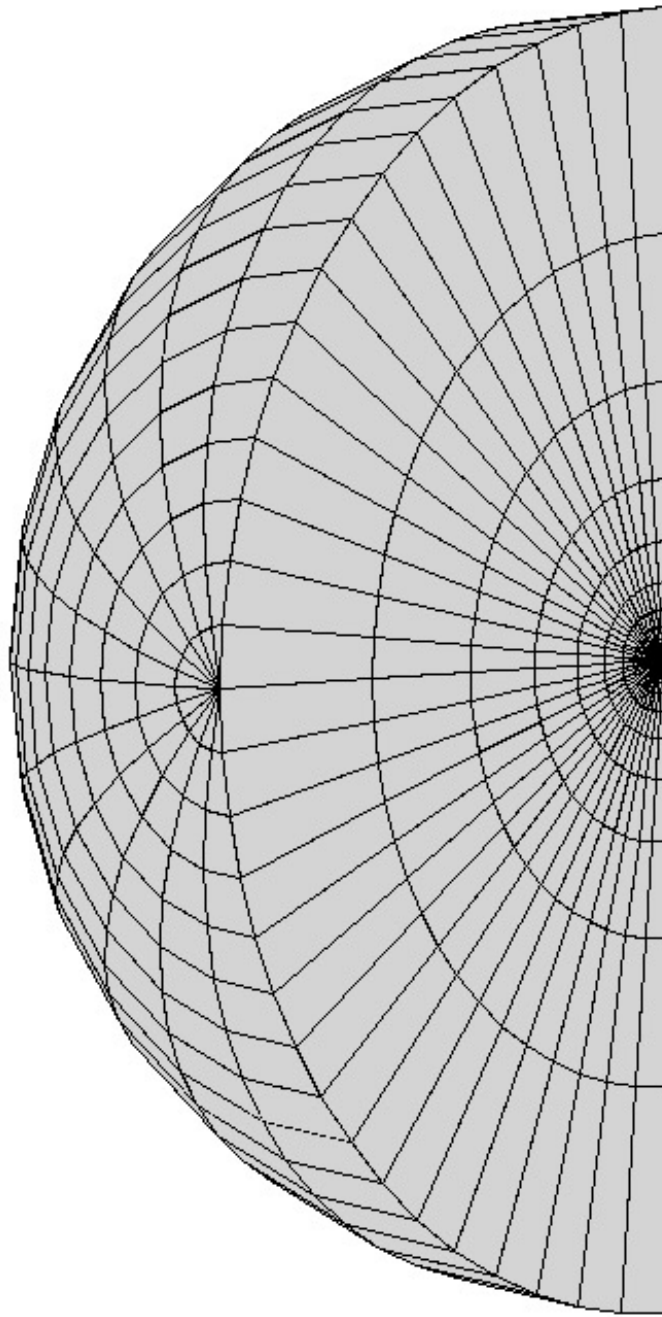


Symmetry plane

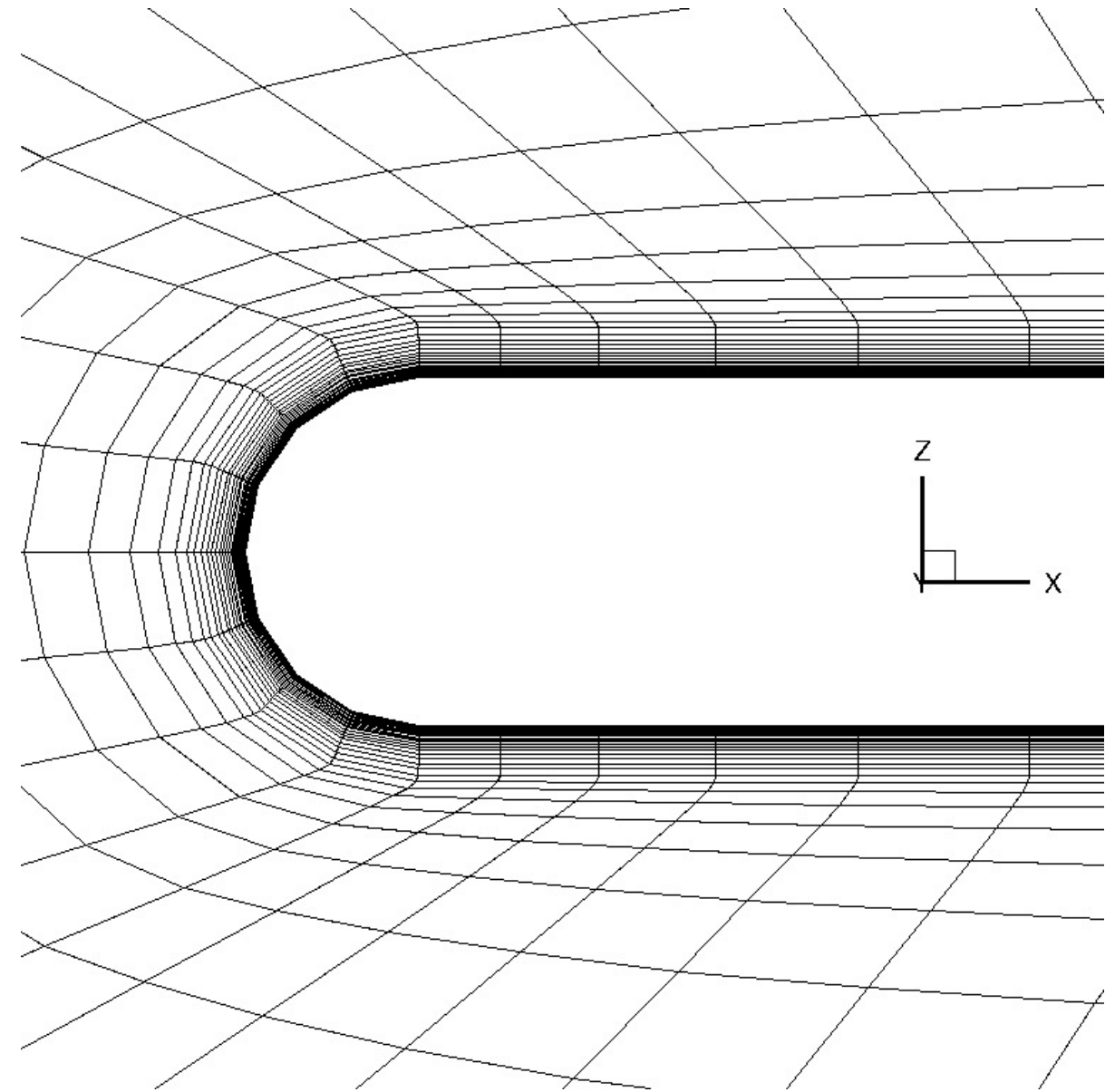


Half: Structured (Hex+prism)

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Symmetry plane



Package



Download hc_v7p5_release_light.tar.gz at TMR website:

HC: NASA Langley TMR website

```
%tar -xvf hc_v7p5_release_light.tar.gz
```

- hcf_hc_v7p5.f90
- hcf_coarsening_v2p1.f90
- readme_release_light.txt
- sample_input , sample_input_coarsen
- sample_input_tets, sample_input_tets_coarsen

```
%source readme_release_light.txt
```

Sample structured grids and tetrahedral grids are generated.

readme_release_light.txt



> Compile the codes

```
gfortran -o hcf_hc hcf_hc_v7p5.f90
```

```
gfortran -o hc_coarsening hcf_coarsening_v2p1.f90
```

> Generate a structured grid

```
hcf_hc < sample_input
```

```
#-----  
# Endianness will be automatically detected, but if you wish,  
# you can specify endianness at compilation: e.g., as follows:  
# gfortran -O2 -fconvert=big-endian -o hcf_hc hcf_hc_v7p5.f90  
# ifort -O2 -convert big_endian -o hcf_hc hcf_hc_v7p5.f90  
#-----
```

> Generate coarser grids

```
hc_coarsening < sample_input_coarsen
```

> Generate a tetrahedral grid

```
hcf_hc < sample_input_tets
```

> Generate coarser grids

```
hcf_coarsening < sample_input_tets_coarsen
```


Input parameters



“sample_input” in hc_v7p5_release_light.tar.gz

```
350000    !Target Reynolds number
1         !Target y-plus value
5         !Element-type: 1=prsm, 2=tets, 3=prsm/tets, 4=prsm/hex, 5=Strct
T         !T = unformatted .ugrid/.ufmt, F = formatted .ugrid/.p3d
100       !Distance to outer boundary (=radius of the outer hemisphere)
48        !Elements along the cylinder
10        !Length of hemisphere-cylinder (apex to base = x2)
16        !Elements along the hemisphere (x=0 to 0.5)
256       !Elements in the radial direction
2         !1: full geometry, 2: a half domain (y > 0 only).
T         !T = Write a boundary grid (Tecplot)
F         !T = Write a volume grid (Tecplot)
T         !T = Write a 'k'-file (required by the coarsening program)
T         !T = Write line files (e.g., for line-relaxation)
T         !T = Write .ugrid file (any type of grid)
T         !T = Write .p3d/.umft and .nmf files (igrid_type = 5 only).
```

Note: Items in red are required for regular coarsening.

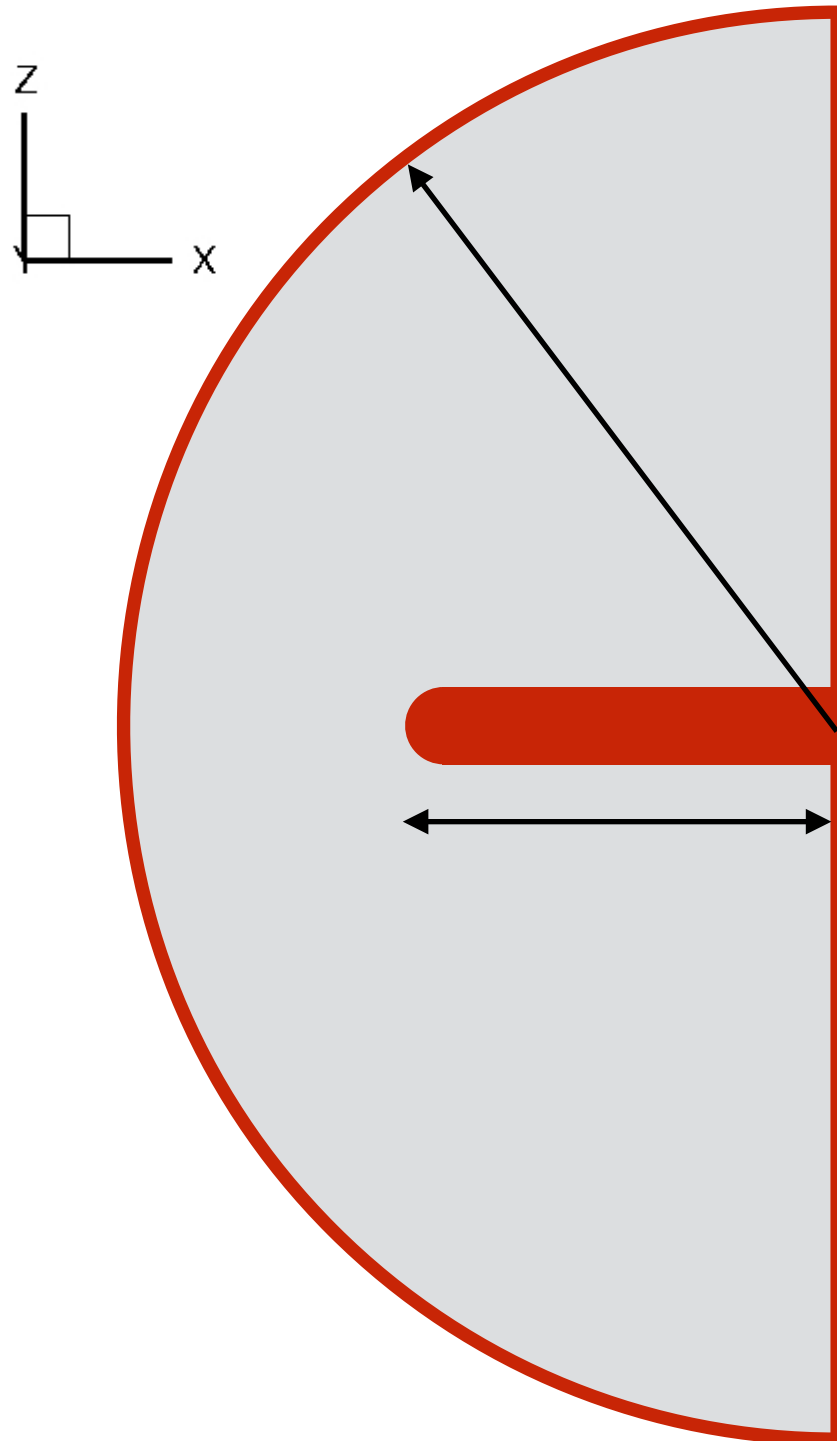
Typical Output



.ugrid/p3d(.ufmt) : Grid file
.mapbc : BC file

.lines_fmt : List of nodes in lines within the BL region
.lines_fmt_all : List of all nodes in lines to the outer boundary.
.k : Structured index file (for coarsening)

Tecplot boundary/volume files.



Hemisphere outer boundary:

Radius of the hemisphere (**input**)

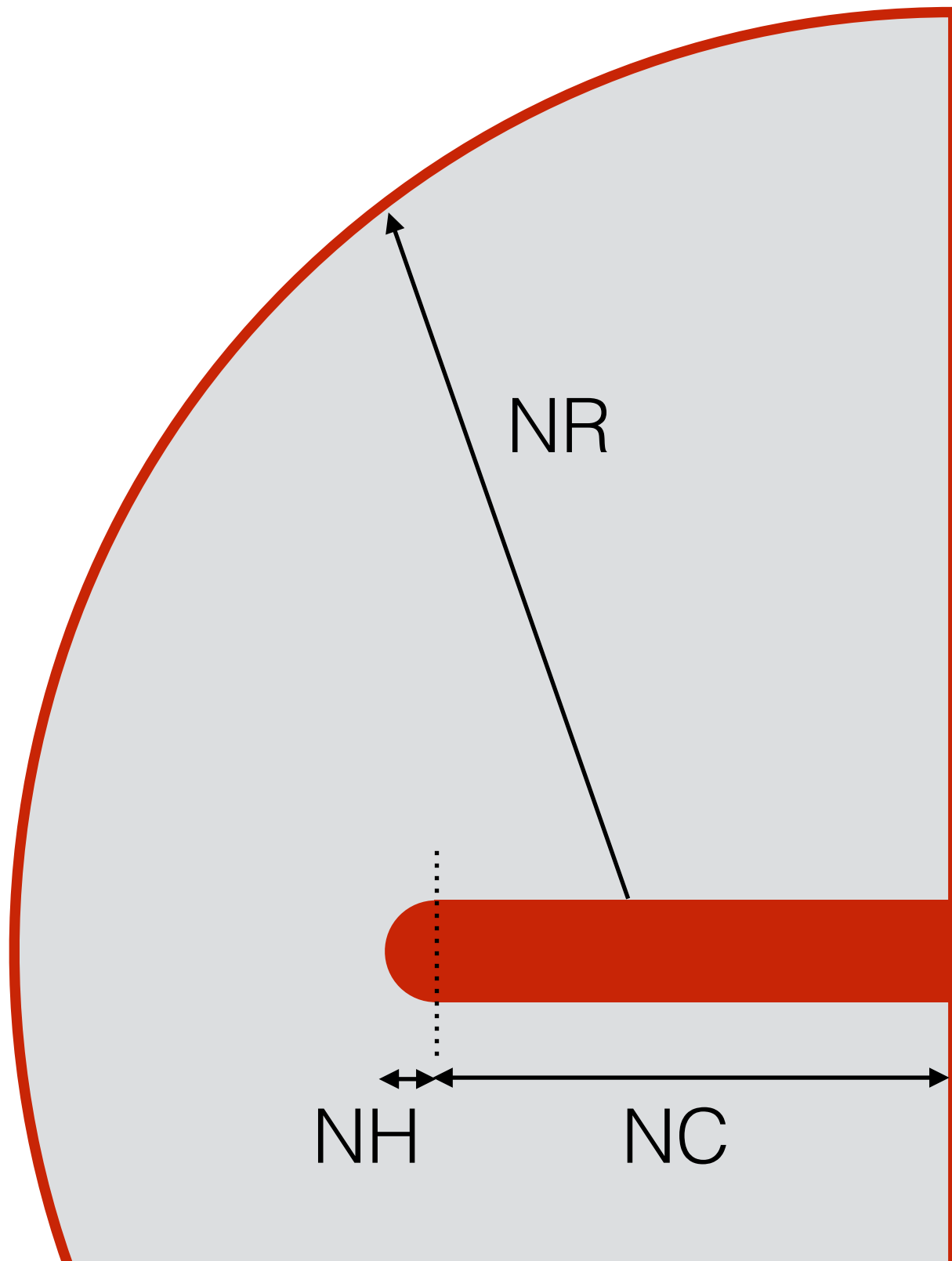
Hemisphere-cylinder:

Apex at the origin

Radius of the hemisphere = 0.5

Length of the HC_{=x-coordinate of the base center} (**input**)

Grid size parameters



Three input parameters.

NC !# of Elements along the cylinder
NH !# of Elements along the hemisphere
NR !# of Elements in the radial direction

Other input parameters.

Re !Target Reynolds number
y_plus !Target y-plus value

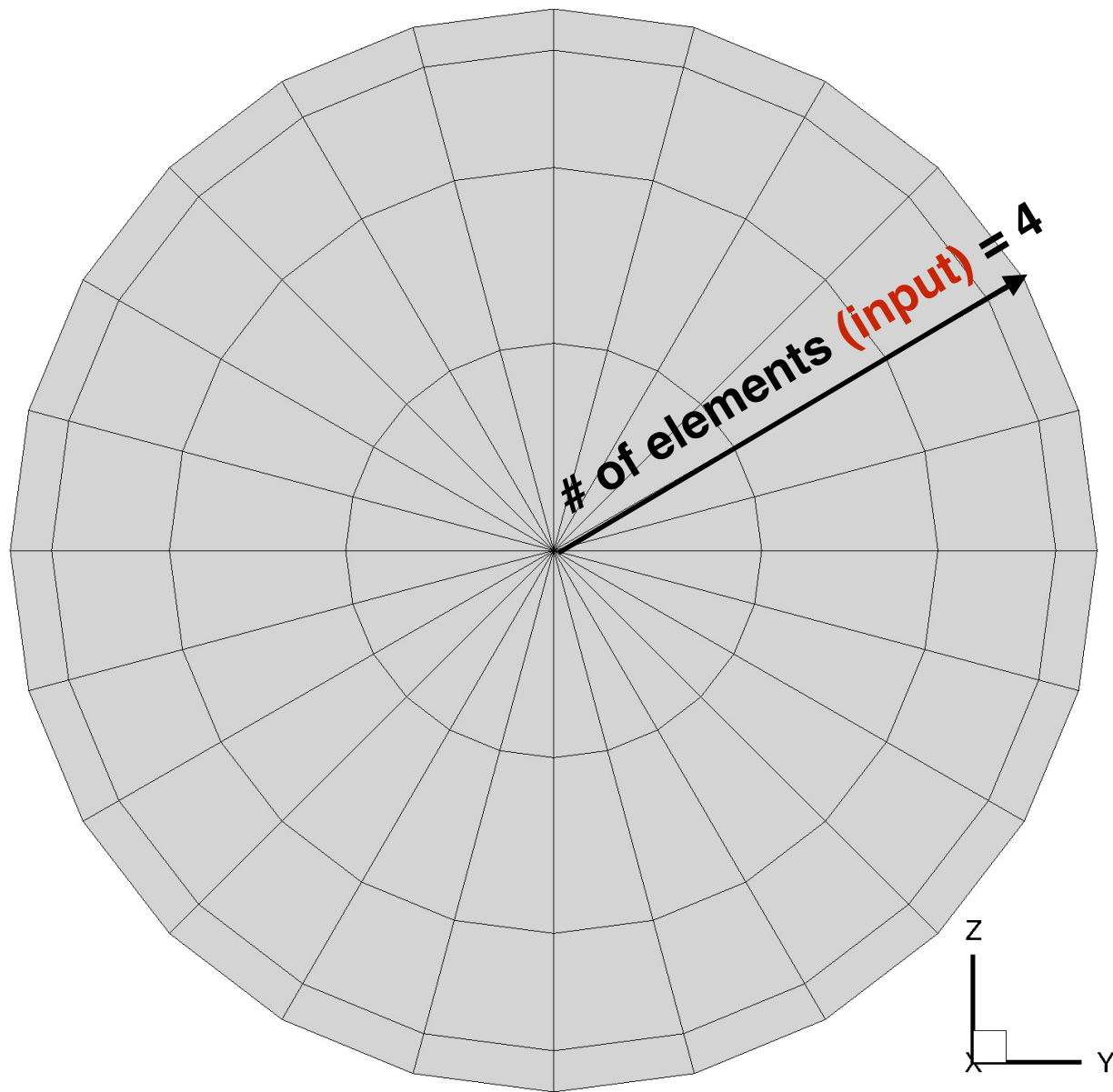
Grid size of the BL region is determined inside the code based on these parameters.

Hemisphere grid



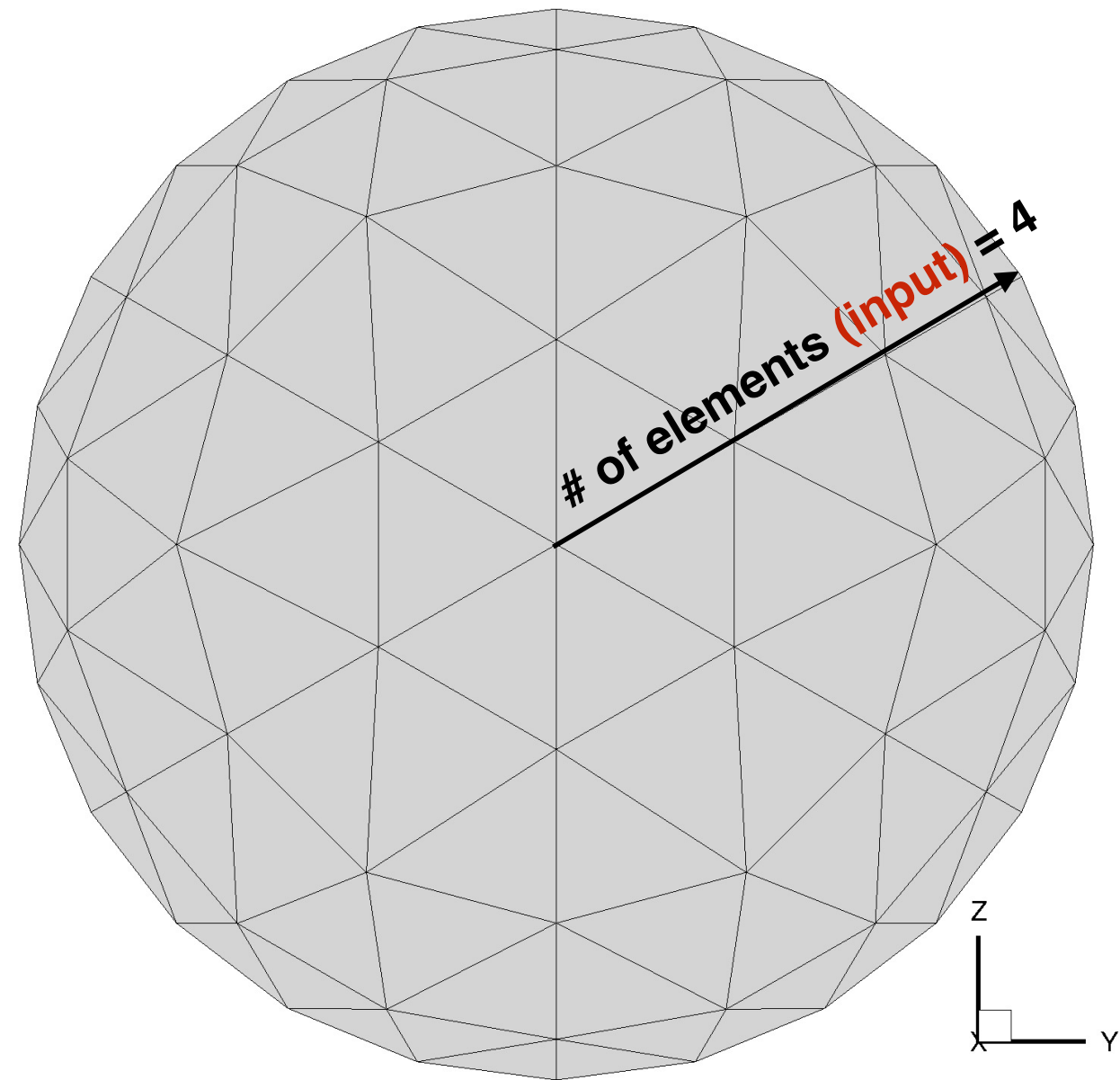
Structured grid

Hex+prisms



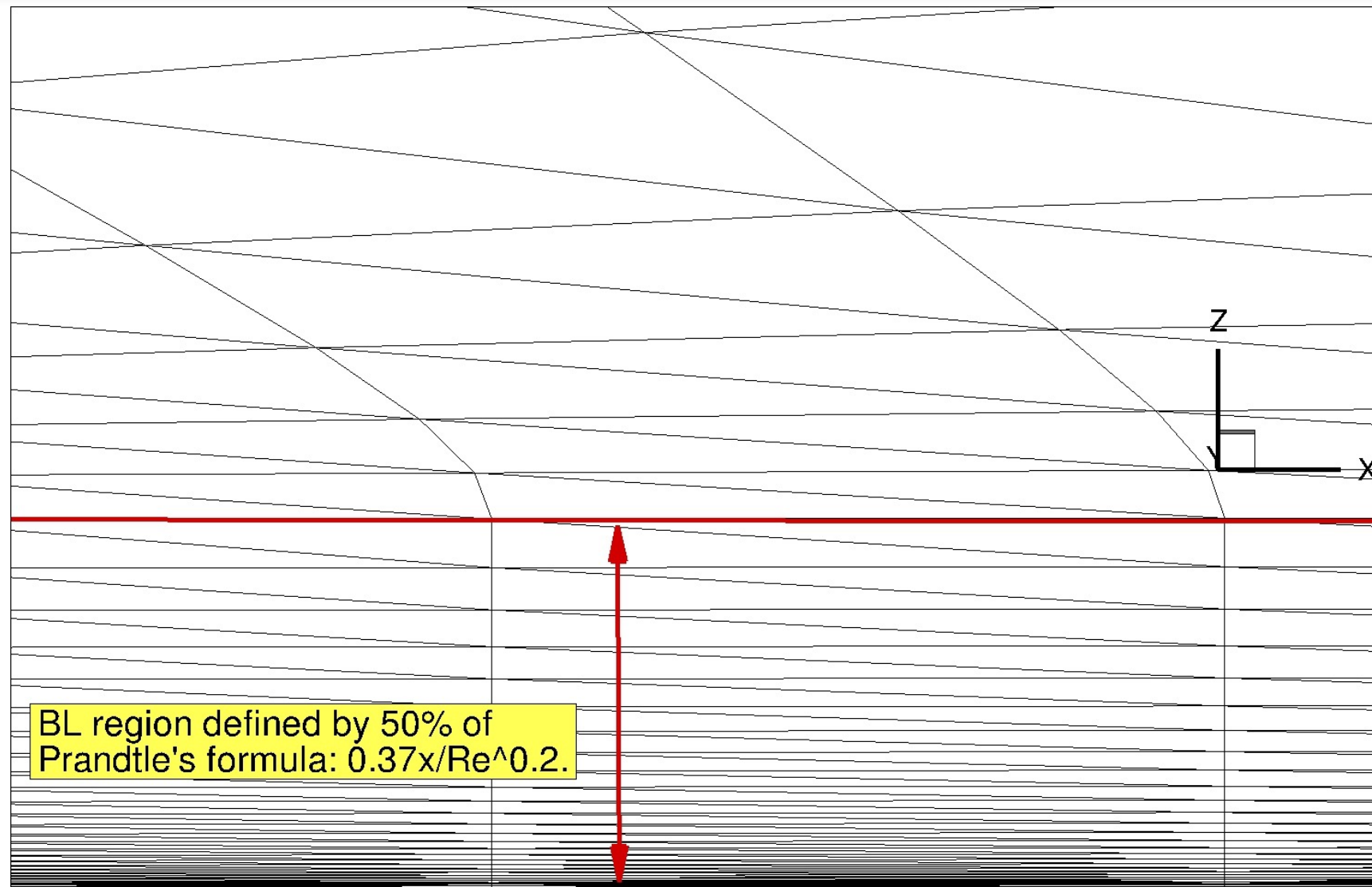
Unstructured grid

Prism/Tetra/Mixed1/Mixed2



Elsewhere, nodes are generated by the same algorithm.

Grid spacing: BL region



Target Re (input)
Target y^+ (input)

BL region thickness is constant over the surface. It has been checked to fully contain boundary layers for intended Reynolds numbers.

y^+ will increase progressively for coarser grids generated by regular coarsening (removing every other node).

First-Off-The-Wall spacing = dr_1 is determined for a target y^+ :
 $dr_1 = \text{target_y_plus} * (\sqrt{2*cf}/Re)$, $cf = 0.026/Re^{(1/7)}$

Nodes in the BL region is determined by a geometric sequence to achieve the outer spacing of 10% of a surface grid spacing.

Grid spacing: outer region

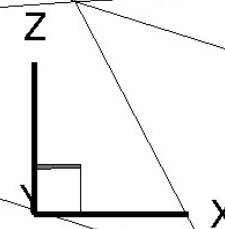


of elements along each radial line outside the BL region is (Input value) - (# of elements in BL region), or if the input is non-positive, it is determined in the code to ensure isotropic elements near the outer boundary and smooth transition from the BL grid.

Nodes are generated with exponential stretching in the direction n1, gradually switching in the direction n2.

n2: Surface node to outer-boundary node

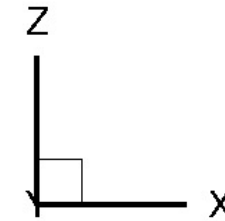
n1: Normal direction



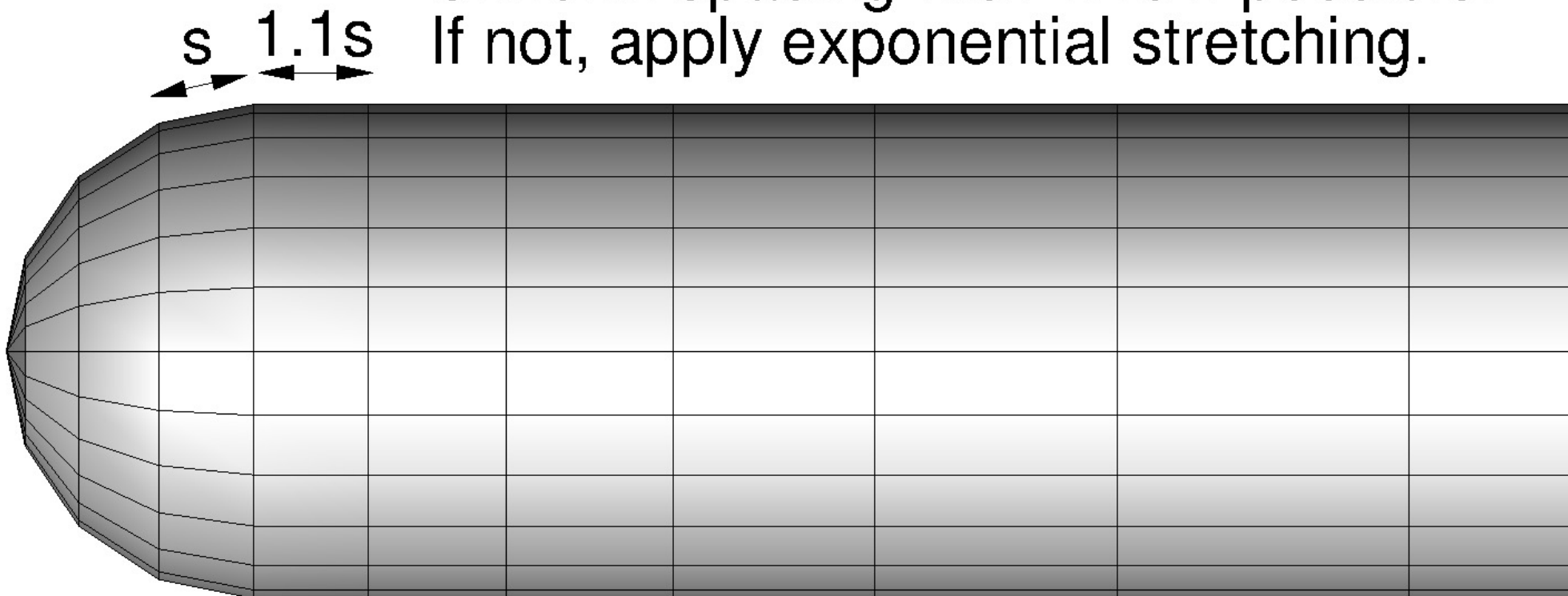
of *total* elements
in the radial direction
(input)

Input value must be large enough, or the code stops and asks you to try again with a suggested value.

Grid spacing: cylinder surface



Uniform spacing with $1.1s$ if possible.
If not, apply exponential stretching.

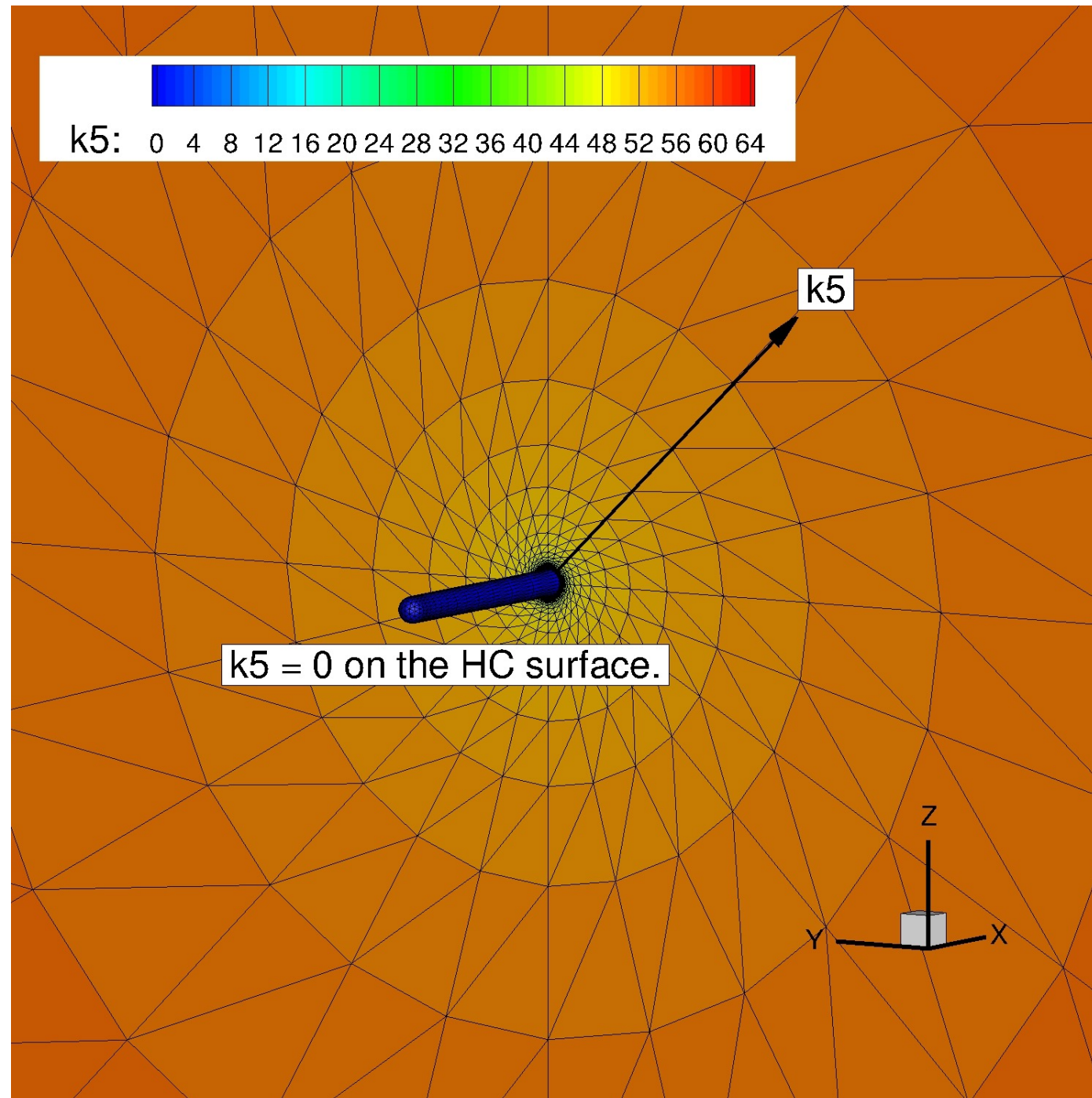


Common Structured Index: k5

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Nodes have structured indices: $(k1, k2, k3, k4, k5)$: **k5 is common to all grids.**

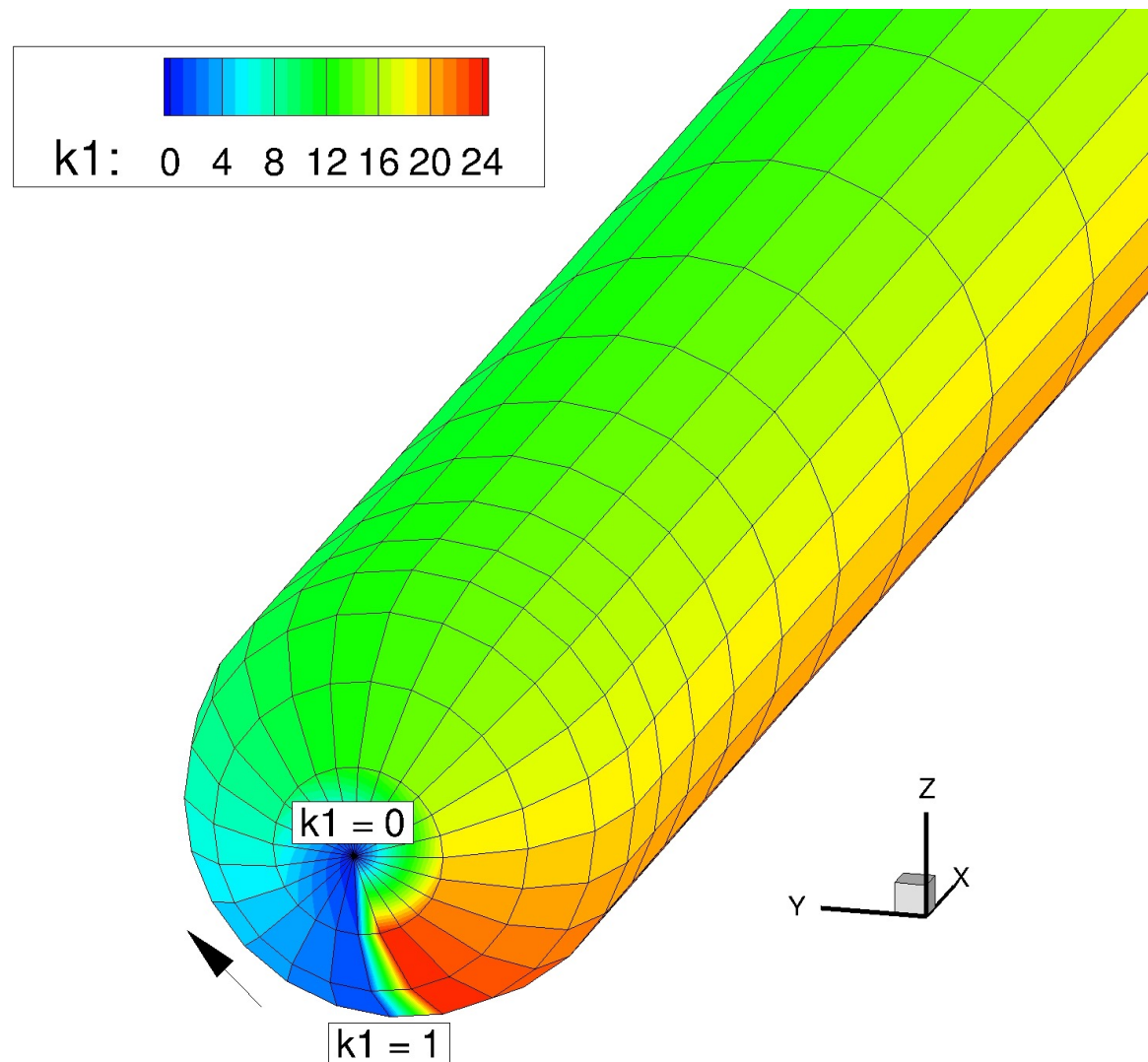


Structured Indices: $k1$, $k2$, $k3$



Structured grid

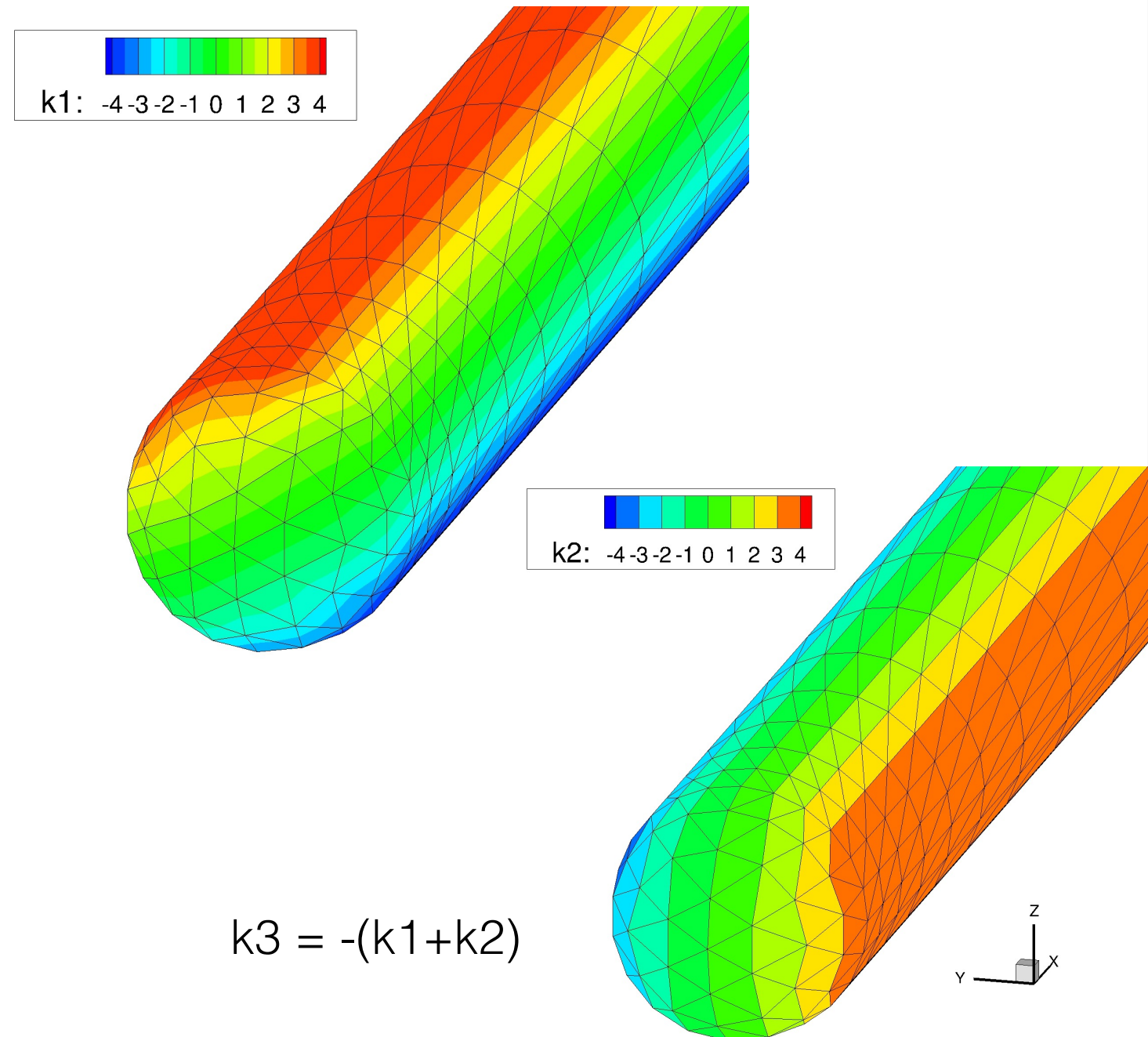
Hex+prisms



$k2$ and $k3$ not used in structured grids.

Unstructured grid

Prism/Tetra/Mixed1/Mixed2

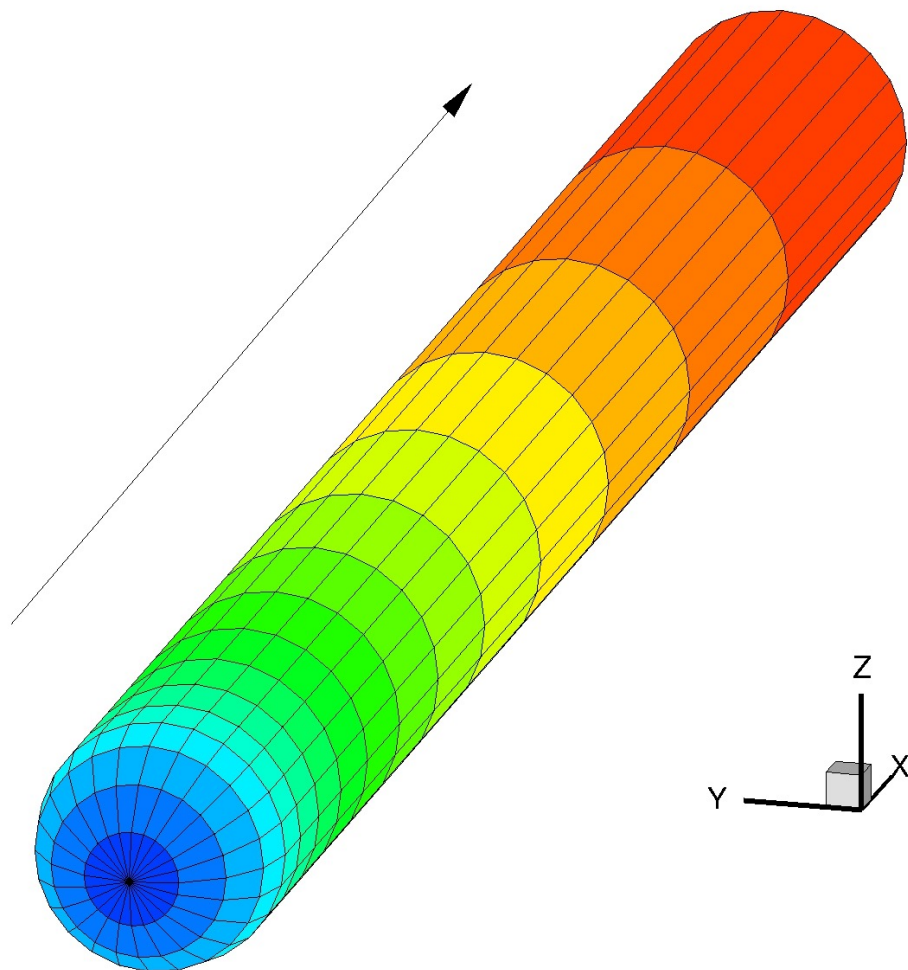
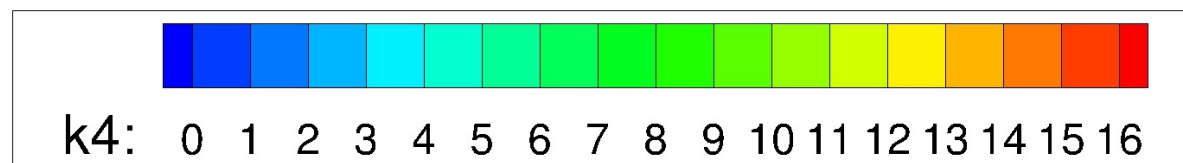


$k3 = -(k1 + k2)$

Structured Index k4

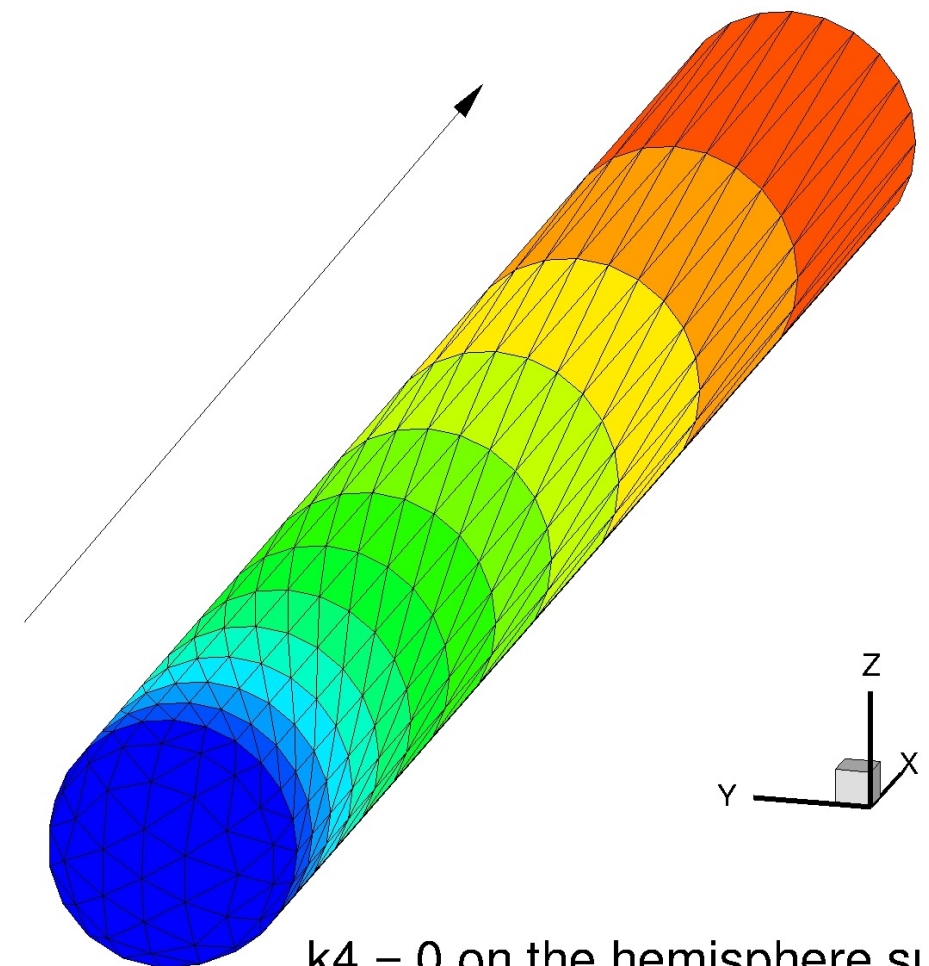
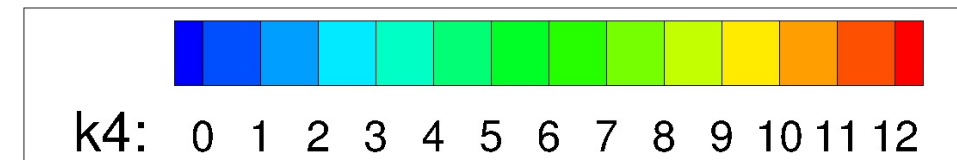
Structured grid

Hex+prisms



Unstructured grid

Prism/Tetra/Mixed1/Mixed2



k4 = 0 on the hemisphere surface.

Lines

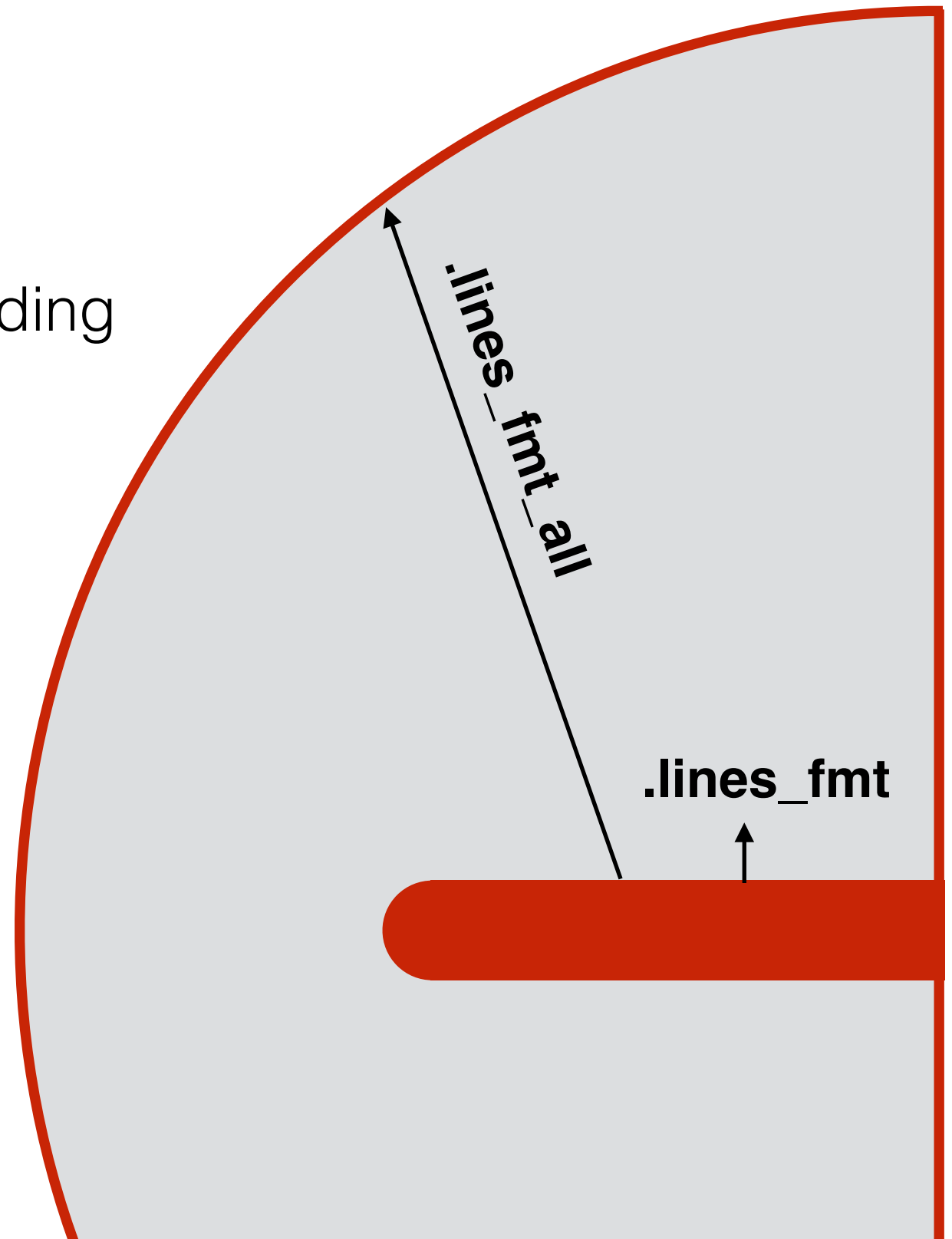
.lines_fmt_all:

List of nodes in the radial grid lines from surface nodes to the corresponding outer-boundary nodes.

.lines_fmt:

List of nodes in lines within the BL region.

can be used for line-agglomeration, line-relaxation.



3D Wing Grids

Topologically equivalent to HC grids

hcf_wing_v3p3.f90

Topologically Equivalent to HC

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Wing grid is generated by mapping a full-geometry HC onto a wing with a specified wing section:

Hemisphere -> Round tip (rotated wing section)

Cylinder -> Wing surface

(Far-field surface mesh is the same between HC and wing.)

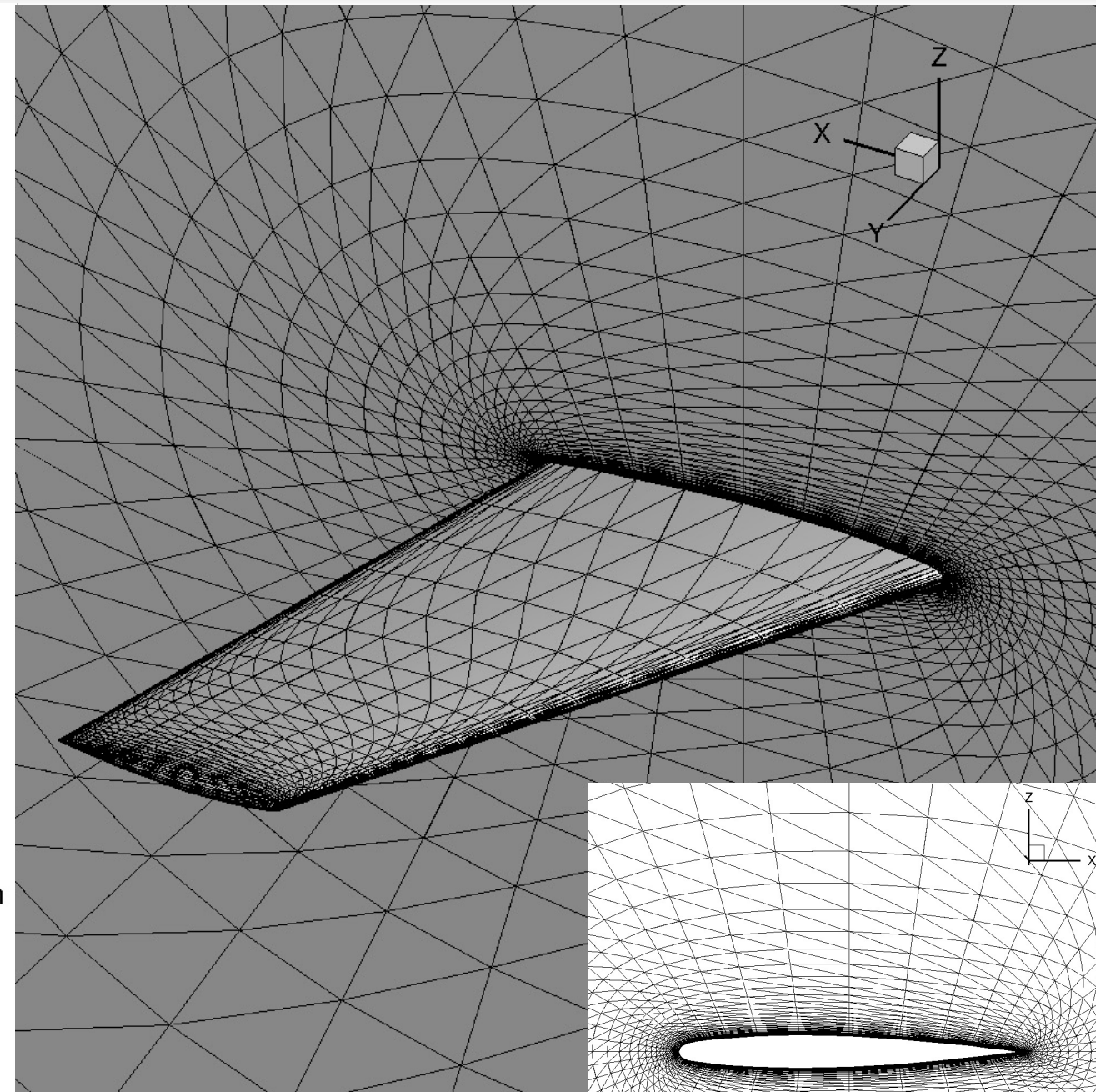
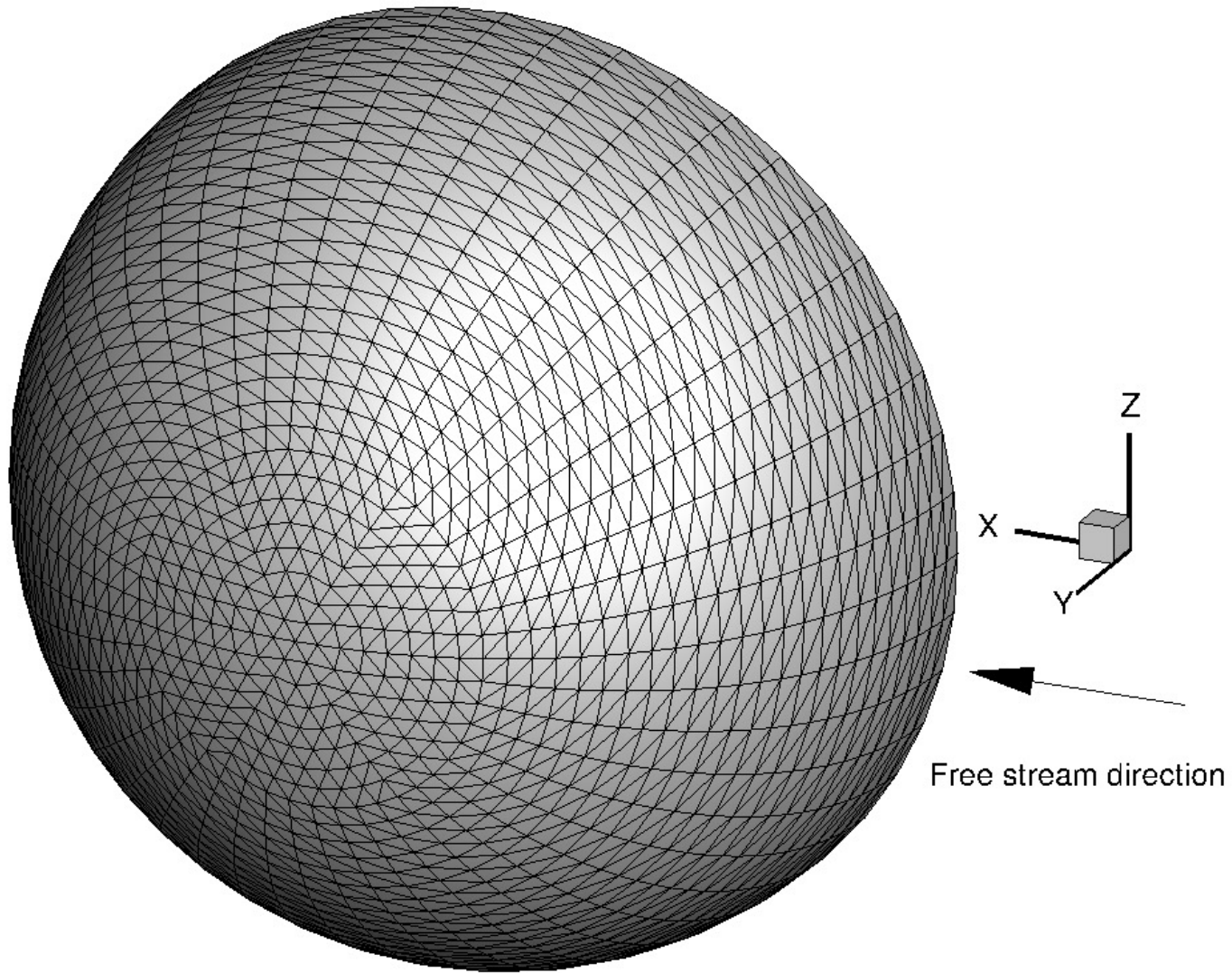
Wing section can be specified by the last two digits of NACA00XX or by a set of discrete points.

The grid is topologically equivalent to a HC grid.

Input values, node generation, line information, and structured indices are the same as the HC grid case.

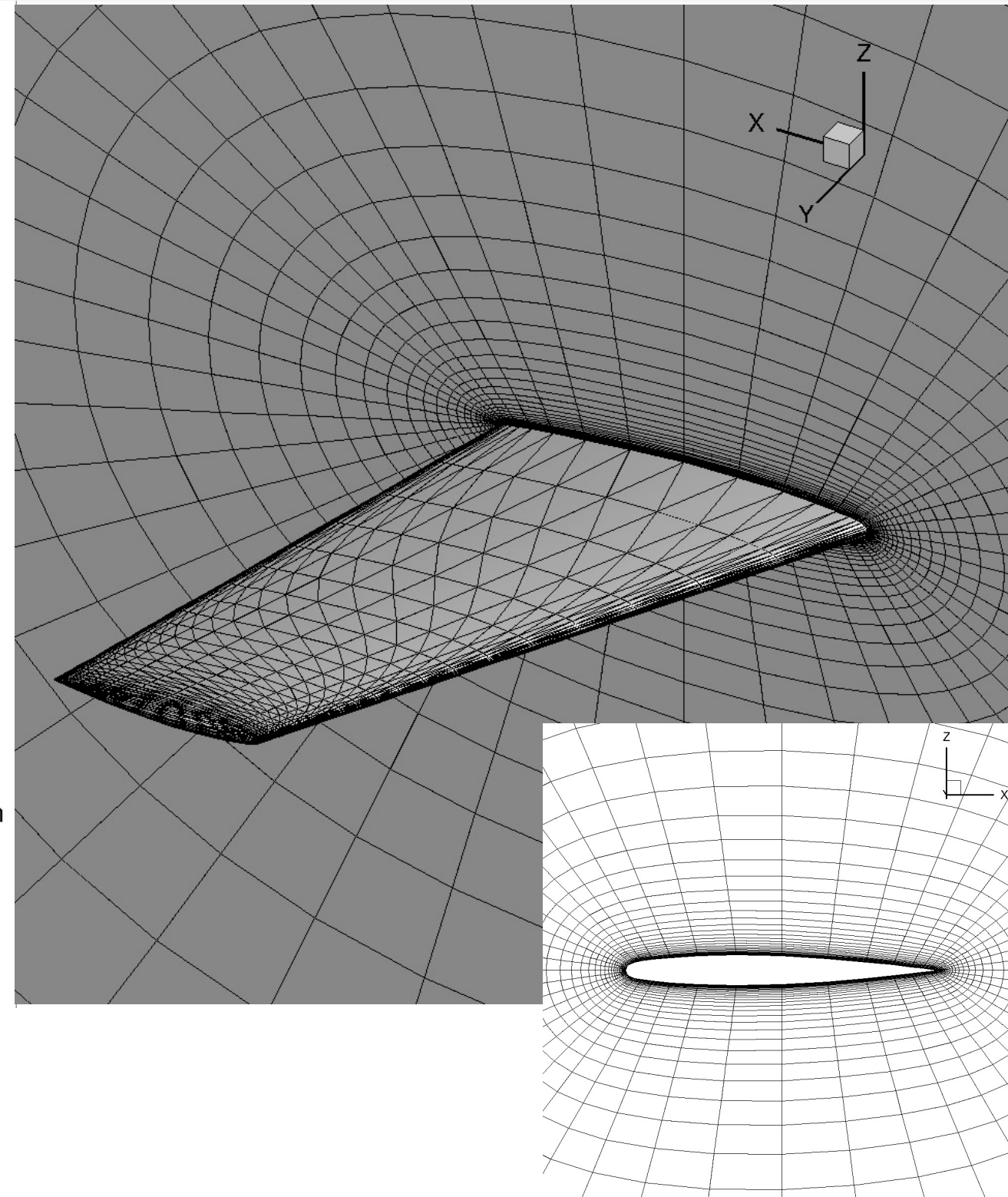
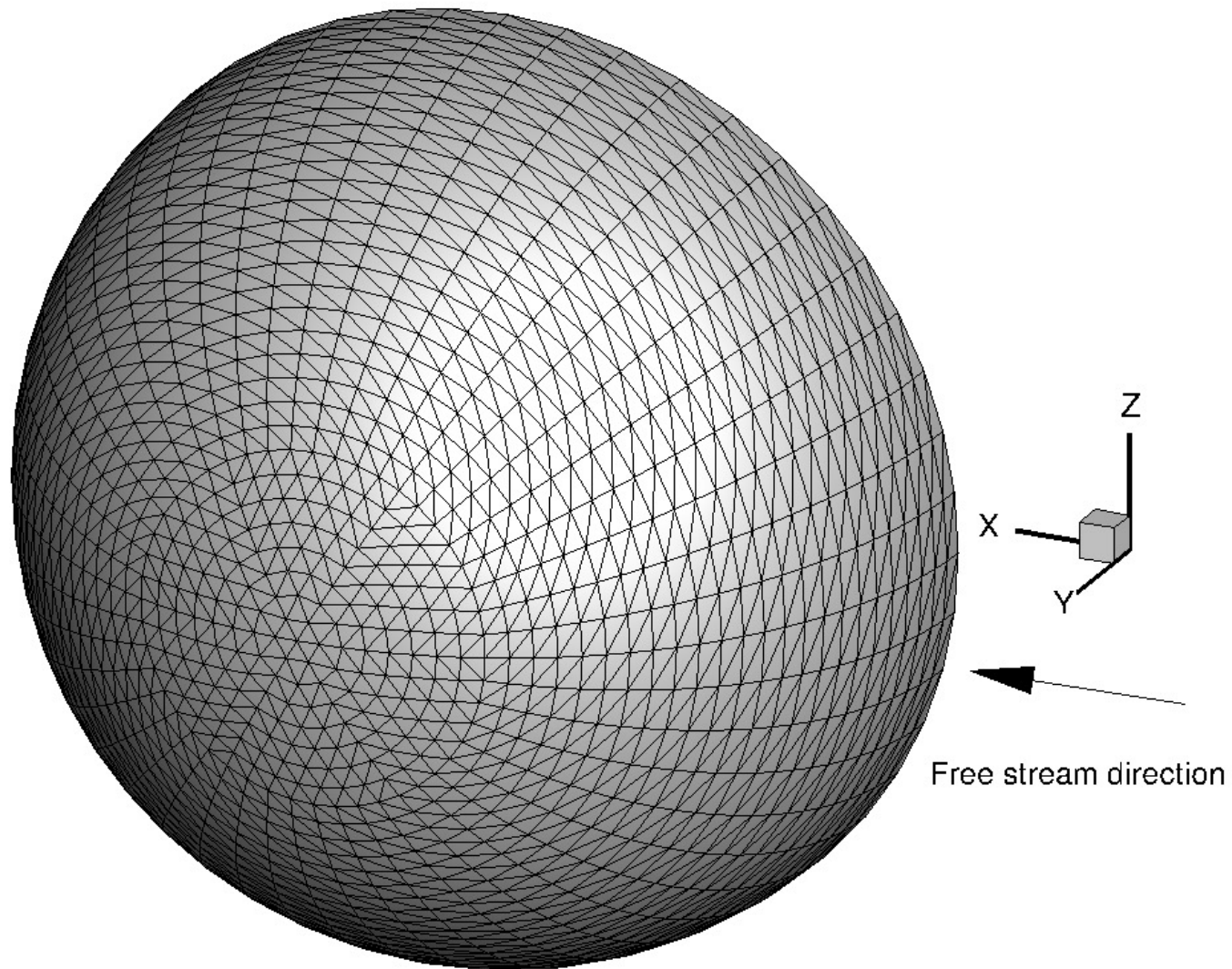
Tetra

Far field boundary (hemisphere shape)

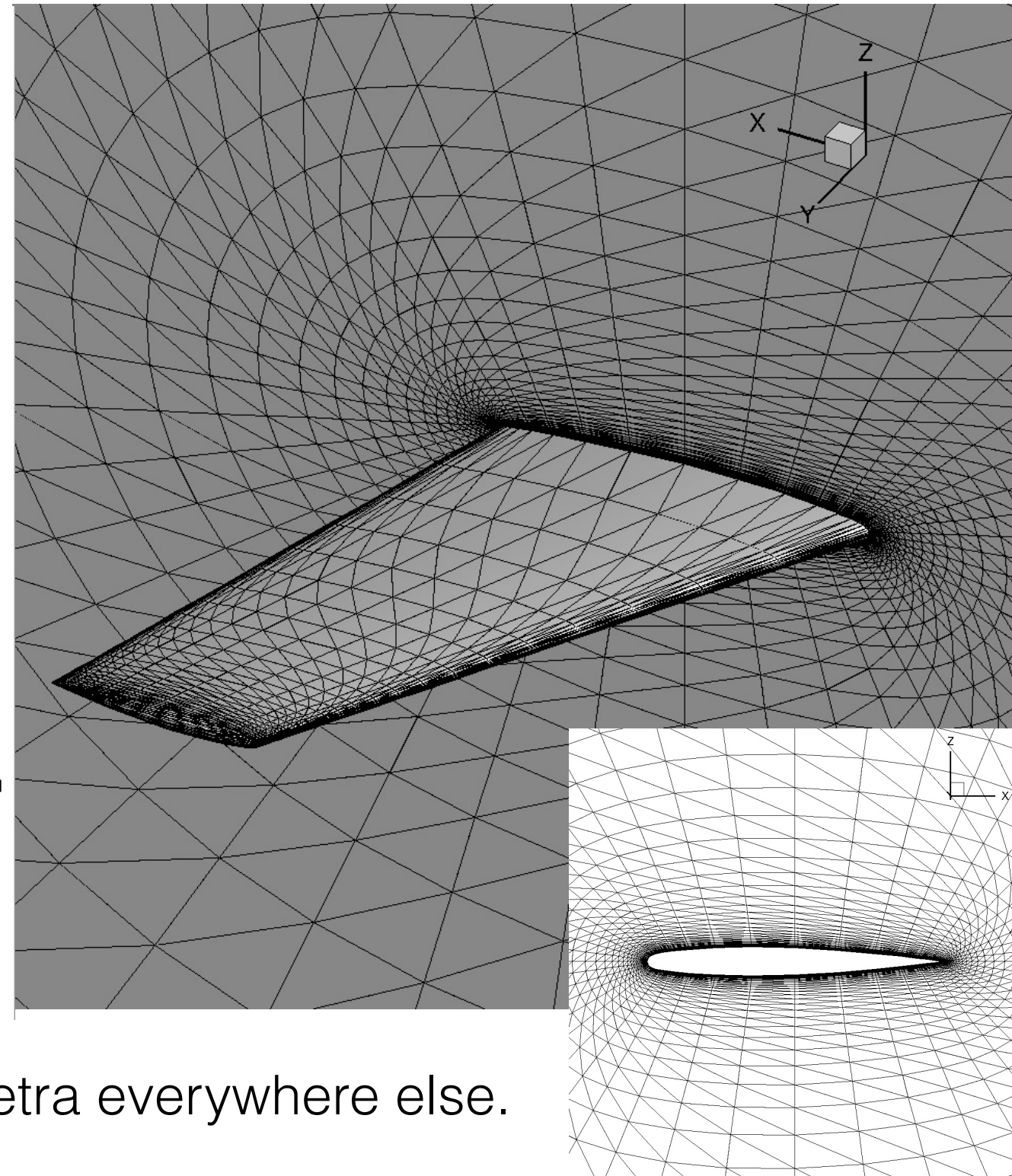
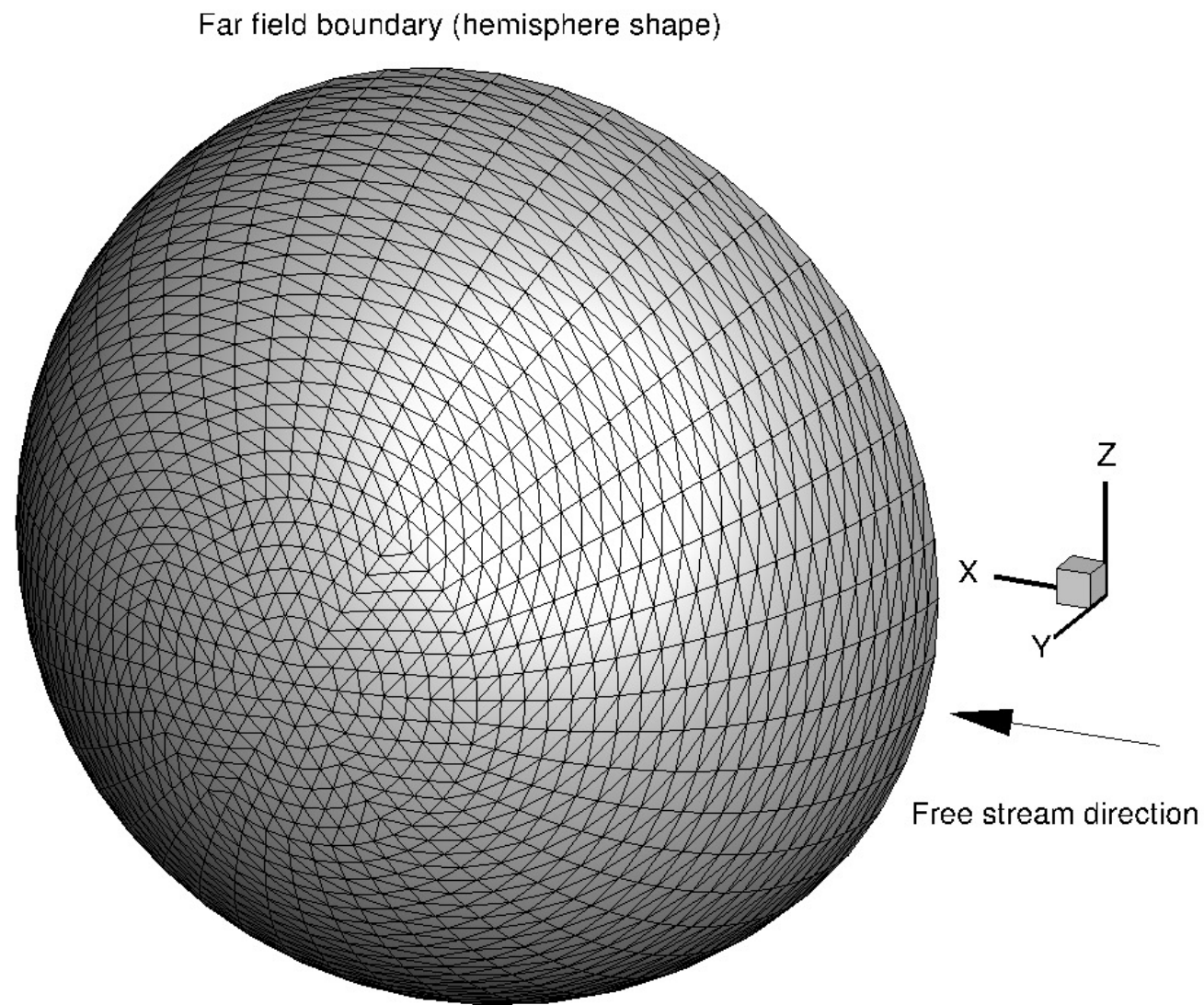


Prism

Far field boundary (hemisphere shape)



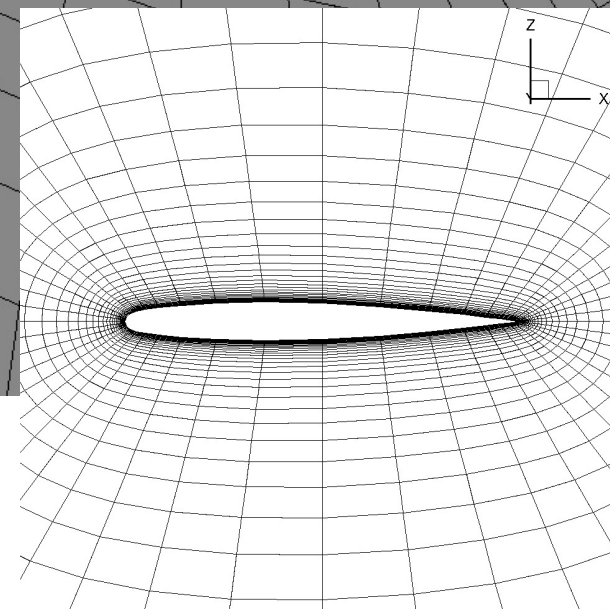
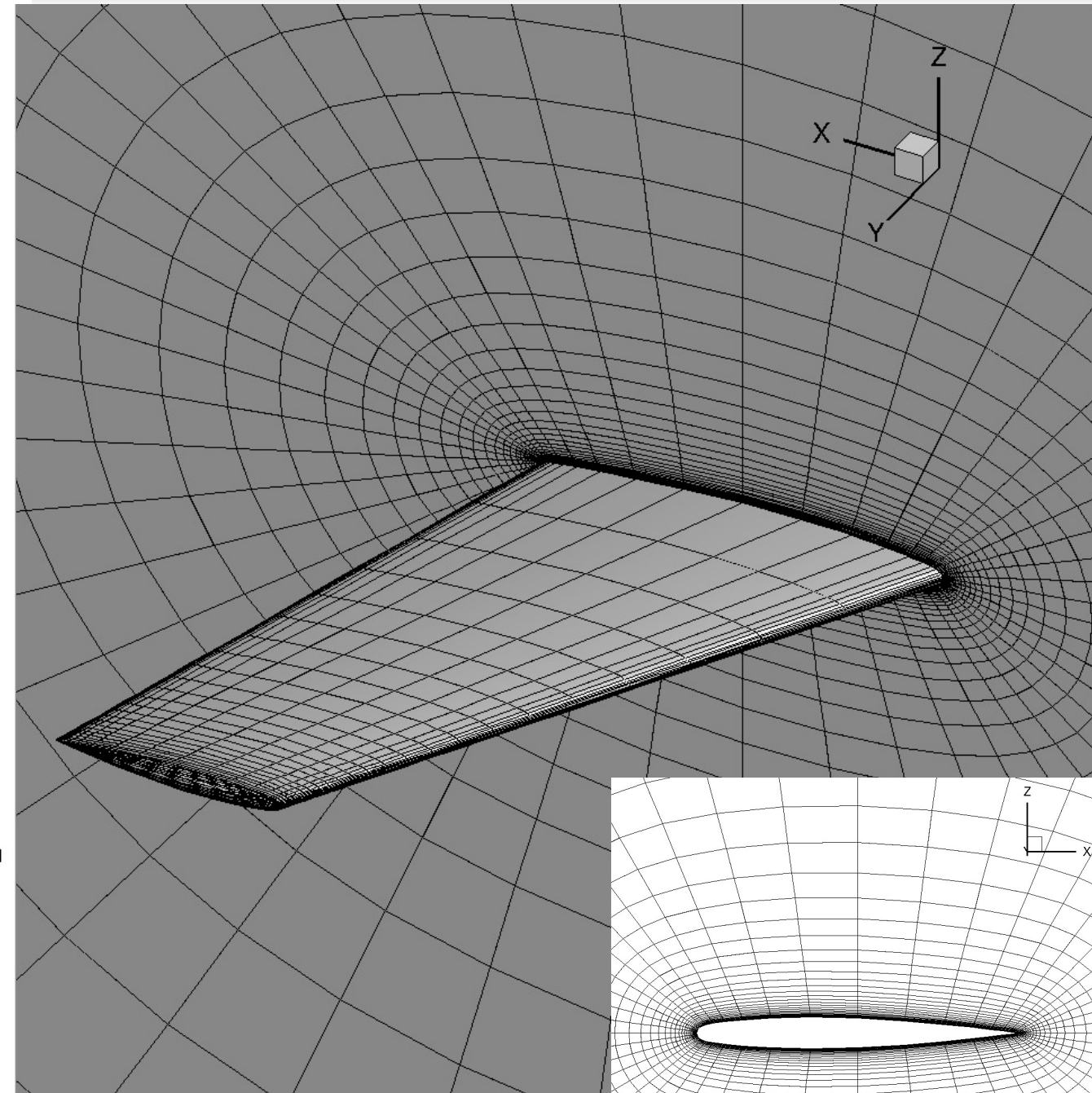
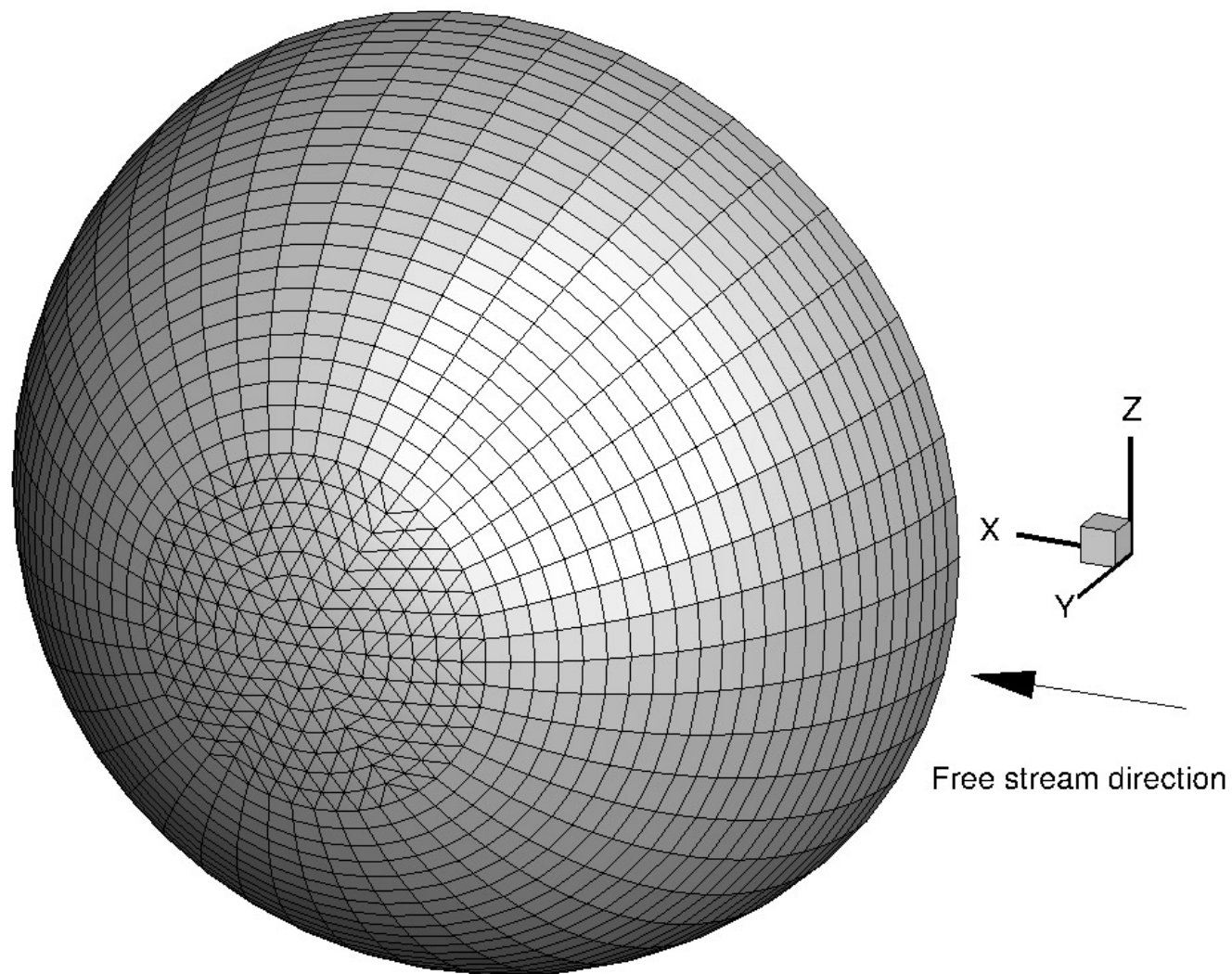
Prism-Tetra



Thin prism layer over the wing surface; tetra everywhere else.

Prism-Hex

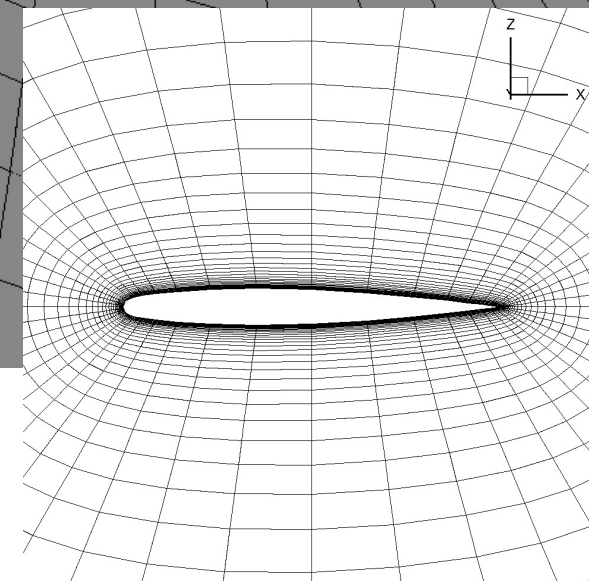
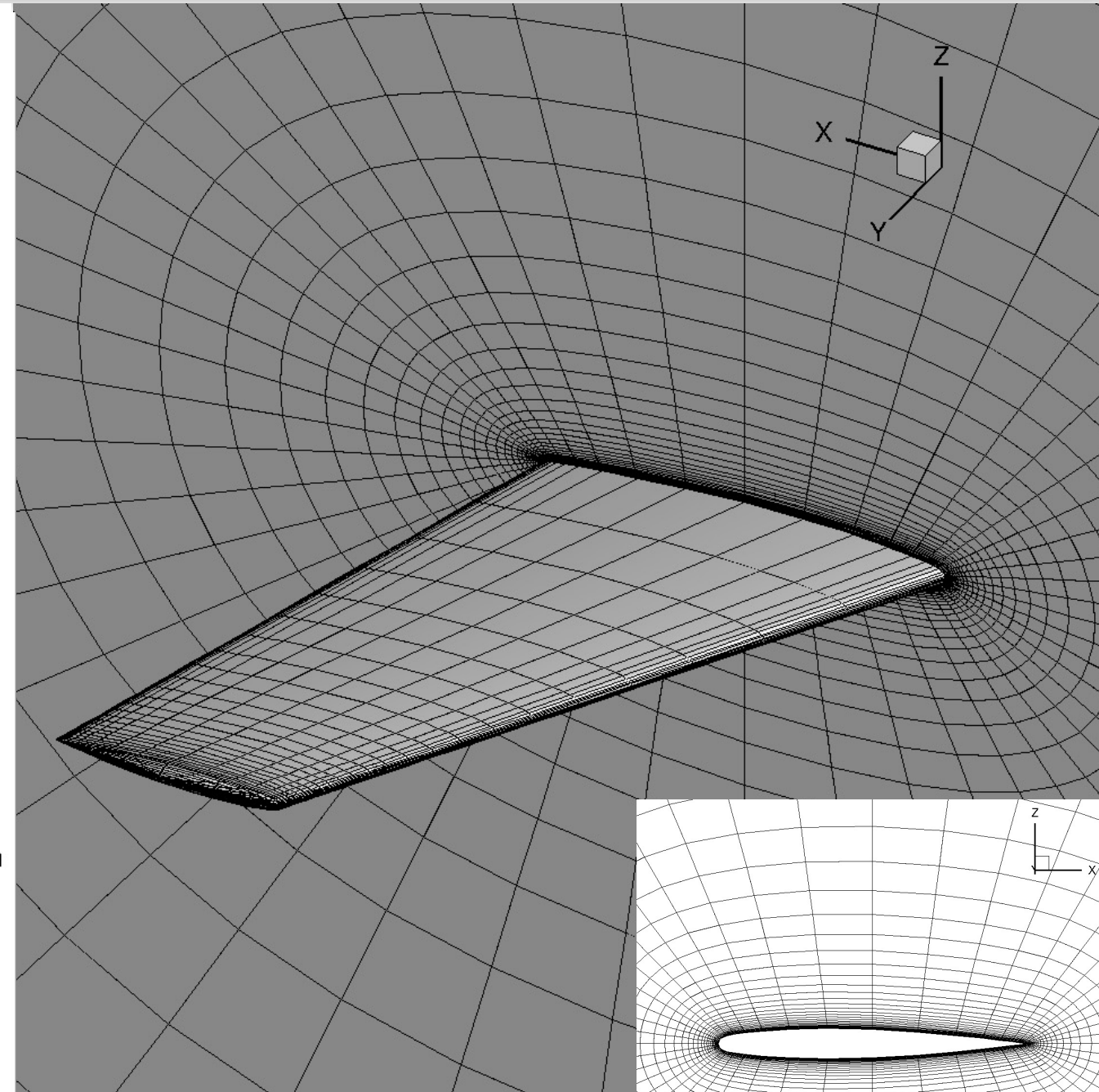
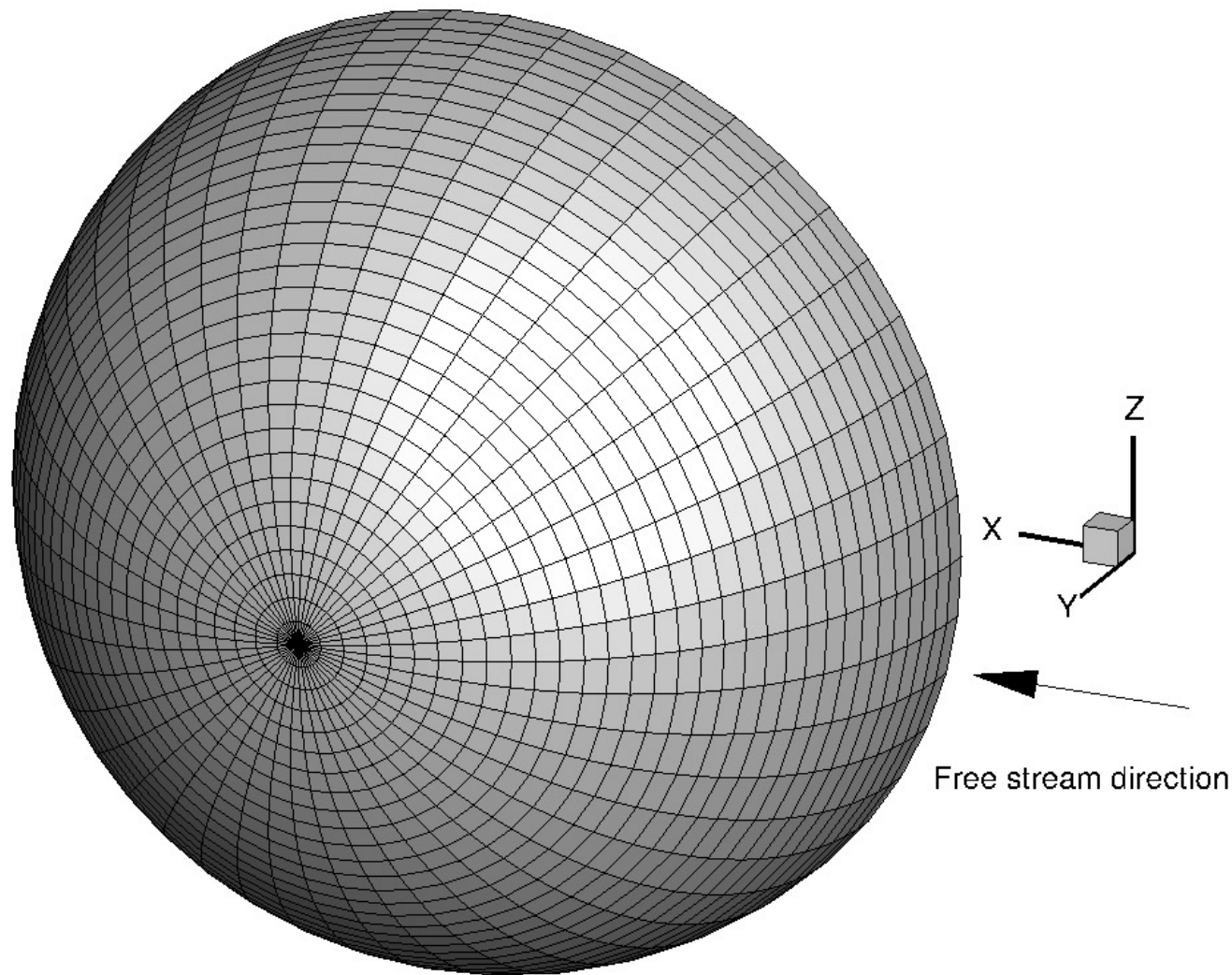
Far field boundary (hemisphere shape)



Prisms over the round tip to farfield; hex everywhere else.

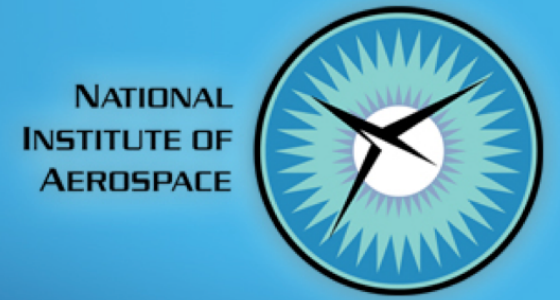
Structured (Hex+prism)

Far field boundary (hemisphere shape)



Prisms are around the center of the round tip; hex are everywhere else.
(Prisms extend to the farfield.)

Package



Download wing_v3p3_release_light.tar.gz at TMR website:

ONERA M6: NASA Langley TMR website

```
%tar -xvf wing_v3p3_release_light.tar.gz
```

- hcf_wing_v3p3.f90
- hcf_coarsening_v2p1.f90
- readme_release_light.txt
- sample_input , sample_input_coarsen
- sample_input_tets, sample_input_tets_coarsen

```
%source readme_release_light.txt
```

Sample structured grids and tetrahedral grids are generated.

readme_release_light.txt



> Compile the codes

```
gfortran -o hcf_wing hcf_wing_v3p3.f90
```

```
gfortran -o hc_coarsening hcf_coarsening_v2p1.f90
```

> Generate a structured grid

```
hcf_hc < sample_input
```

```
#-----  
# Endianness will be automatically detected, but if you wish,  
# you can specify endianness at compilation: e.g., as follows:  
# gfortran -O2 -fconvert=big-endian -o hcf_wing hcf_wing_v3p3.f90  
# ifort -O2 -convert big_endian -o hcf_wing hcf_wing_v3p3.f90  
#-----
```

> Generate coarser grids

```
hc_coarsening < sample_input_coarsen
```

> Generate a tetrahedral grid

```
hcf_hc < sample_input_tets
```

> Generate coarser grids

```
hcf_coarsening < sample_input_tets_coarsen
```

Input parameters



“sample_input” in wing_v3p2_release_light.tar.gz

14.6e6	!Target Reynolds number based on the root chord (=1 in the grid).
1	!Target y-plus value
2	!Element-type: 1=prsm, 2=tets, 3=prsm/tets, 4=prsm/hex, 5=Strct
T	!T = unformatted .ugrid/.ufmt, F = formatted .ugrid/.p3d
0	!airfoil_type: =0 discrete airfoil data, =1 NACA00XX
om6_wing_section_sharp.dat	!Datafile for the discrete airfoil data ; not used if airfoil_type = 1
0	!Last two digits of NACA00XX (15 for NACA0015); not used if airfoil_type = 0
0.5625159852668158	!Taper ratio = (tip chord)/(root chord)
29.9990	!Swept angle of LE line in degrees (not too large, please: e.g., < 40).
100	!Distance to outer boundary (=radius of the outer hemisphere)
24	!# of Elements along the semi-wingspan
1.476017976219800	!b = semi-span: per unit root chord.
8	!# of Elements along the rounded tip divided by 2.
88	!# of Elements in the radial direction (from wing to farfield)
2	!wing_side: =1 for left wing, =2 for right wing.
0.0	!root_le_x: = x coordinate of the LE at the root.
T	!T = Write a boundary grid (Tecplot)
F	!T = Write a volume grid (Tecplot)
T	!T = Write a 'k'-file (required by the coarsening program)
T	!T = Write line files (e.g., for line-relaxation)
T	!T = Write .ugrid file (any type of grid)
T	!T = Write .p3d/.umft and .nmf files (igrid_type = 5 only)

Note: Items in red are required for regular coarsening.

Currently, the code accepts only a symmetric airfoil.

NACAXX (airfoil_type = 1):

Specify the last two digits as input parameter.

Discrete data (airfoil_type = 0):

Provide a file containing the points that define the half-thickness of a desired airfoil (the number of points, followed by a list of x-coordinates, and then a list of y-coordinates).

Specify the filename as input parameter. The code constructs cubic splines from the discrete data, and uses it to generate a grid.

Example of an OM6 wing section is provided:

See “om6_wing_section_sharp.dat” [AIAA J. Vol. 54, No. 9, September 2016.](#)

Example: ONERA M6 Wing

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Airfoil data: “om6_wing_section_sharp.dat”

Data taken from **AIAA J. Vol. 54, No. 9, September 2016.**

Input parameters: All input given for unit root-chord length.

Target Re = 14.6e6

Leading Edge Sweep = 29.9990 degrees

Taper ratio = 0.5625159852668158

semi-span = 1.47601797621980

Output



.ugrid/p3d(.ufmt) : Grid file
.mapbc : BC file

.lines_fmt : List of nodes in lines within the BL region
.lines_fmt_all : List of all nodes in lines to the outer boundary.
.k : Structured index file (for coarsening)

Tecplot boundary/volume files.

Regular Coarsening for HC/Wing grids

hcf_coarsening_v2p1.f90

Regular coarsening



HC/Wing grids can be regularly coarsened by using the coarsening program (included in each package):

`hcf_coarsening_v2p1.f90`

It will coarsen a target grid (generated by the HC/Wing grid generation code) by removing every other node. It continues to coarsen grids until it is not possible.

Required Input files



Four files must be available for a target fine grid:

xxx.1.ugrid

xxx.1.lines_fmt

xxx.1.lines_fmt_all

xxx.1.k

where xxx = 'hc_tetra', 'wing_tetra', etc.

Make sure these files are generated when you generate a target fine grid by the HC/Wing grid generation program.

Coarsening level



Coarsening level is determined by the input parameters used in the HC/Wing grid generation program.

Key input parameters.

NC !# of Elements along the cylinder
NH !# of Elements along the hemisphere (apex to shoulder)
NR !# of Elements in the radial direction

of Coarse Grids

$$\# \text{ of coarse grids} = \min(m4, m1, m5)$$

—Structured:

Full geometry

$$k4=NC+NH = 2^{m4}, \quad k1=6 \cdot NH = 2^{m1}, \quad k5=NR = 2^{m5}$$

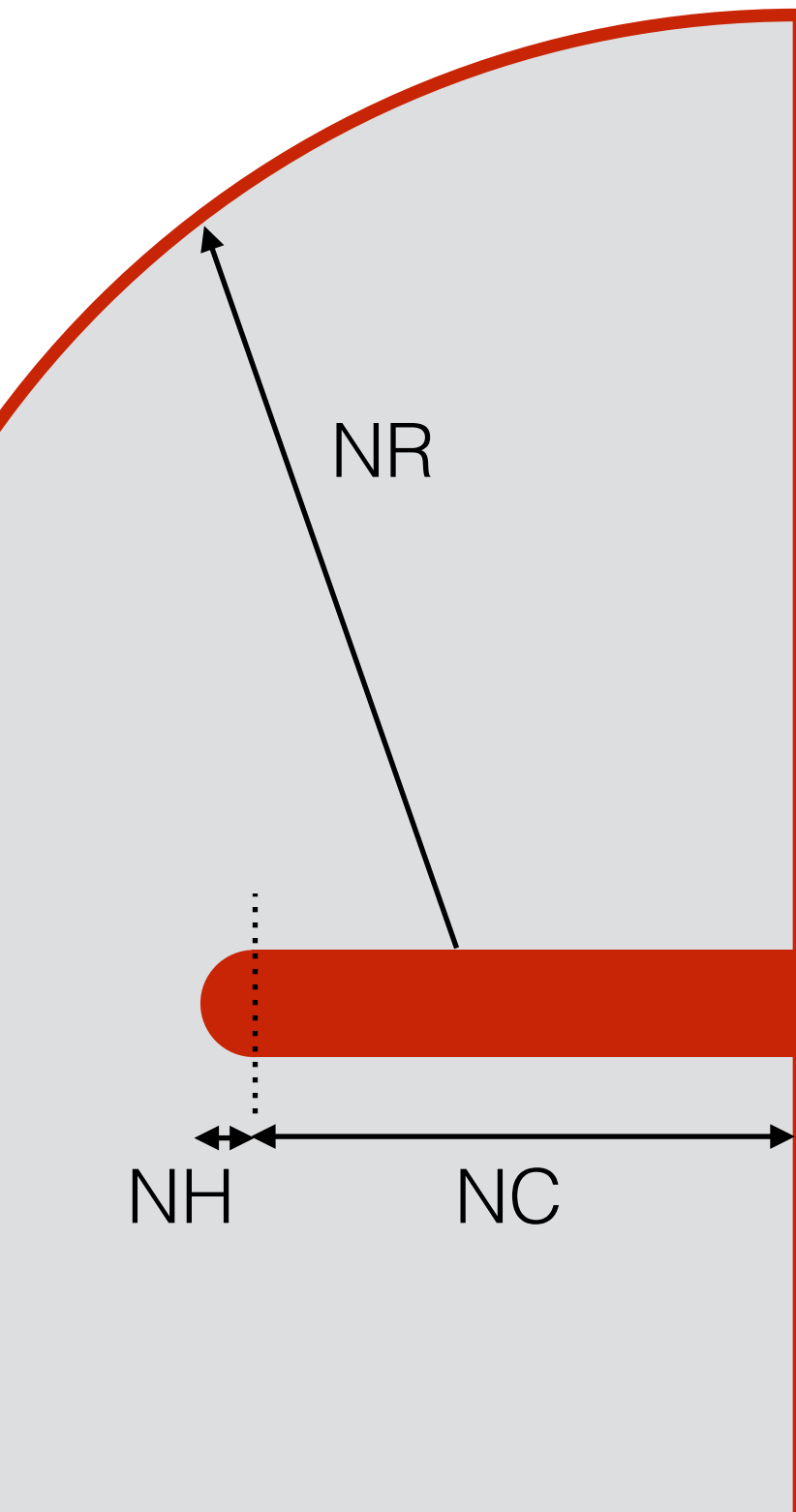
Half geometry

$$k4=NC+NH = 2^{m4}, \quad k1=3 \cdot NH = 2^{m1}, \quad k5=NR = 2^{m5}$$

—Unstructured:

$$k4=NC = 2^{m4}, \quad k1=NH = 2^{m1}, \quad k5=NR = 2^{m5}$$

HC/Wing grid generation code prints out the maximum possible coarsening level on screen as soon as it reads input parameters.



Input parameters



Sample input parameters for hcf_coarsening_v2p1.f90

Example for “hc_tetra”:

```
hc_tetra !project name: = 'hc_prism', 'hc_tetra', 'hc_mixed', 'hc_mixed_ph', 'hc_strct'  
                                'wing_prism', 'wing_tetra', 'wing_mixed', 'wing_mixed_ph', 'wing_strct'  
  
T      !T=unformatted .ugrid (F=formatted)  
F      !T=Tet2Prism on coarse grids, F=Keep tets  
T      !T=Write a boundary grid (Tecplot)  
F      !T=Write a volume grid (Tecplot)
```

It can generate all-prism coarse grids for a tetrahedral grid.
The code continues coarsening until it is not possible.



.ugrid/p3d(.ufmt) : Grid files for all coarse grids.
.mapbc : BC file
.lines_fmt : List of nodes in lines within the BL region
.lines_fmt_all : List of all nodes in lines to the outer boundary.
.k : Structured index file (for coarsening)

.prolong_nc : Inter-grid interpolation information
.prolong_nc_seq: List of nested nodes
.midp : List of edge 'mid-point' coordinates.
can be used to create a P2 grid

Tecplot boundary/volume files.

Coarsening example (HC)

Structured grids:

Half geometry

Finest grid generated by

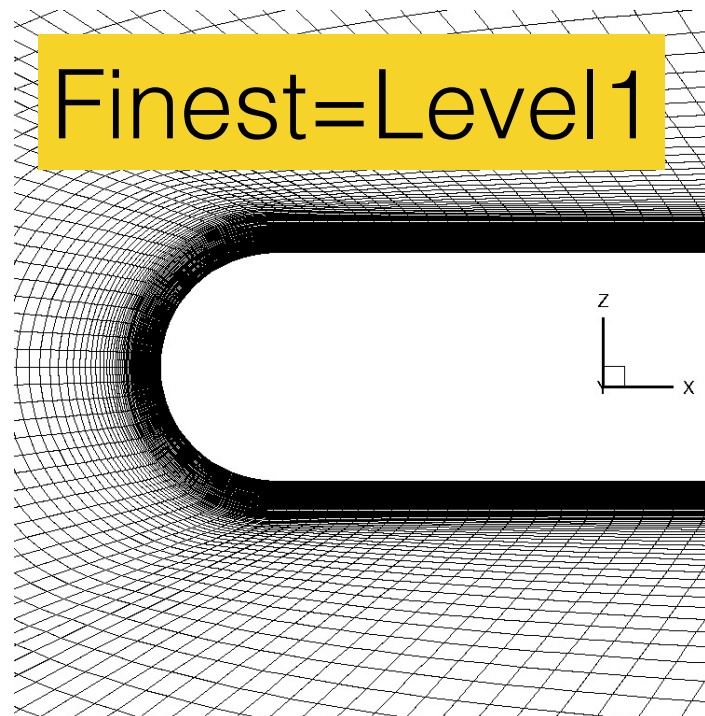
NC = 48

NH = 16

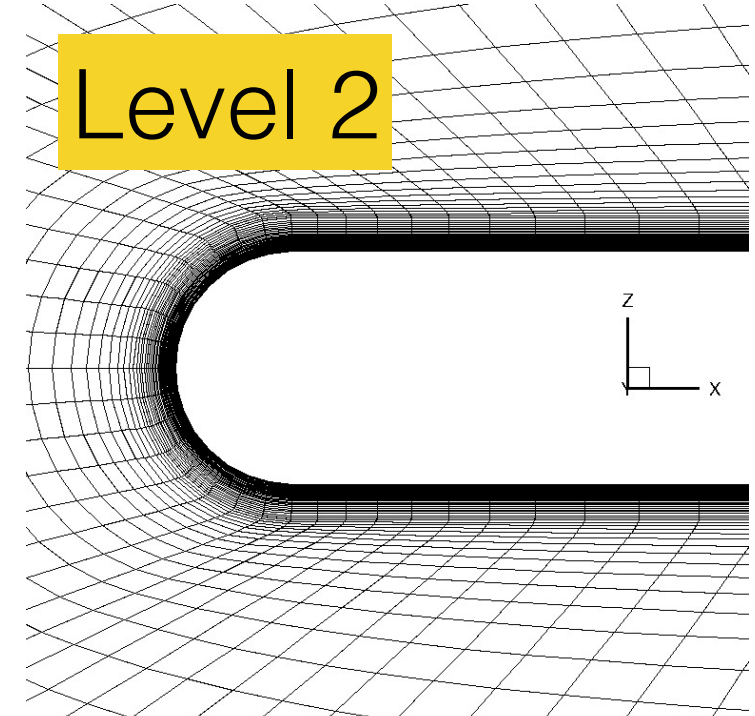
NR = 256

See “sample_input”

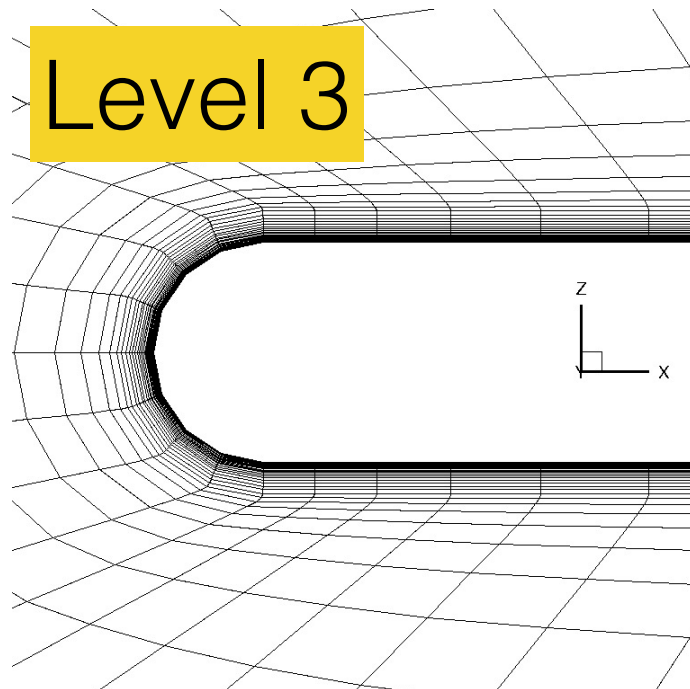
Finest=Level1



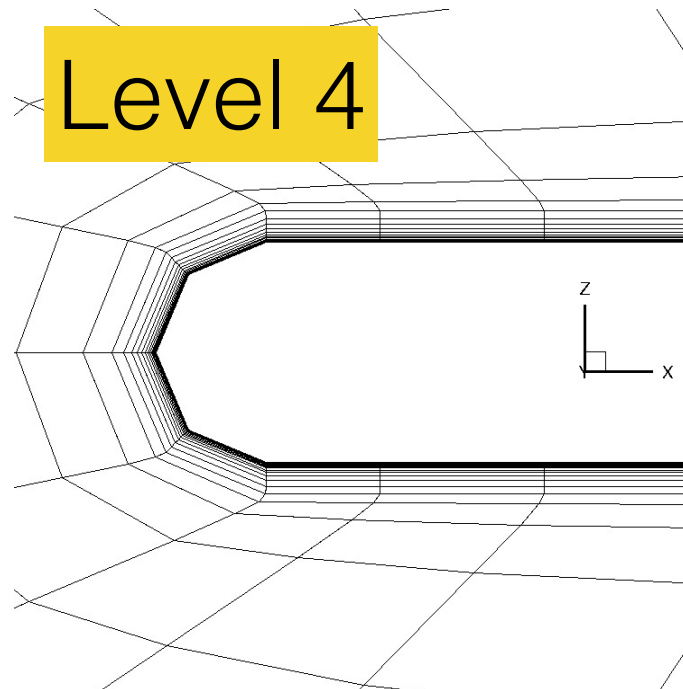
Level 2



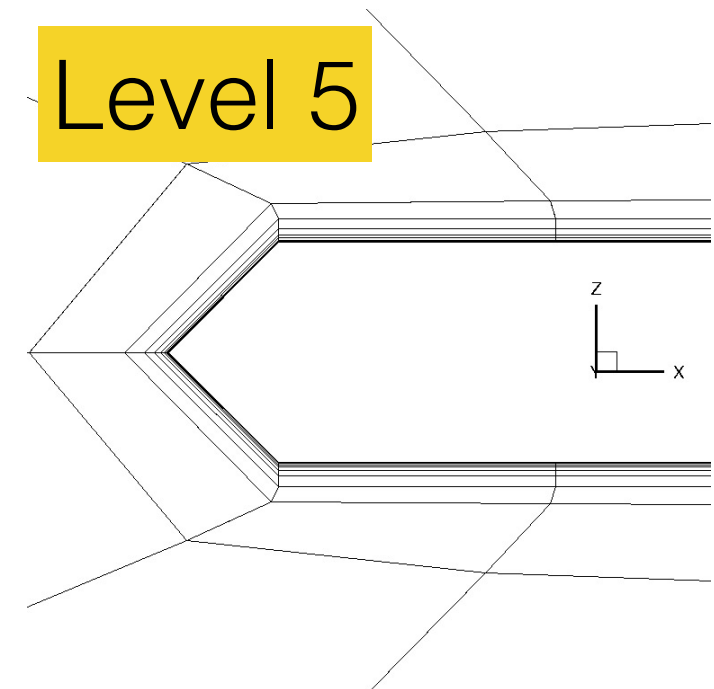
Level 3



Level 4



Level 5



Coarsening example (Wing)



Tetra grids:

Finest grid generated by

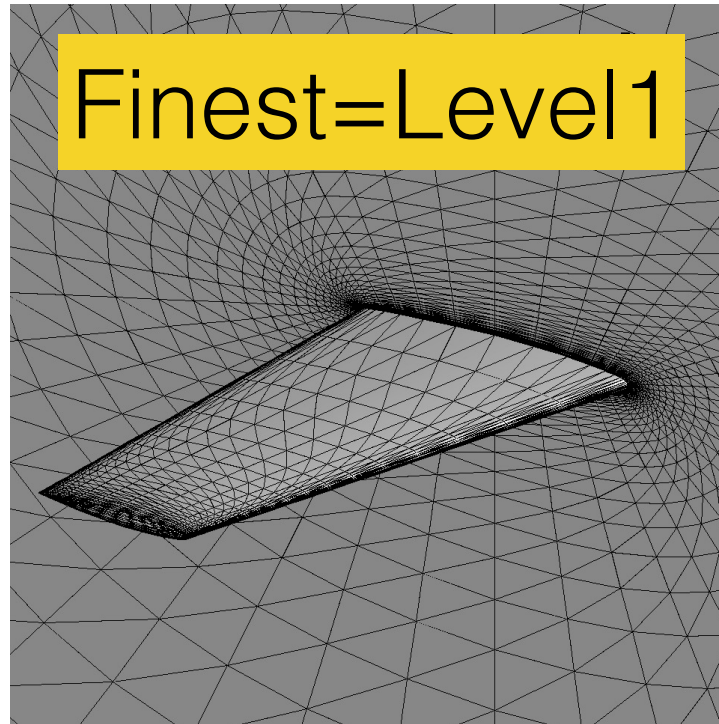
NC = 24

NH = 8

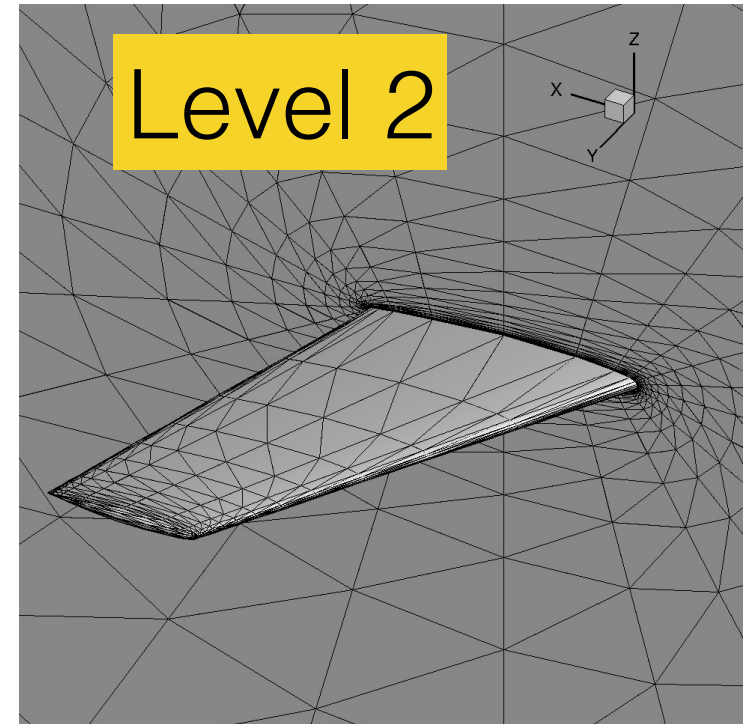
NR = 88

See “sample_input_tets”

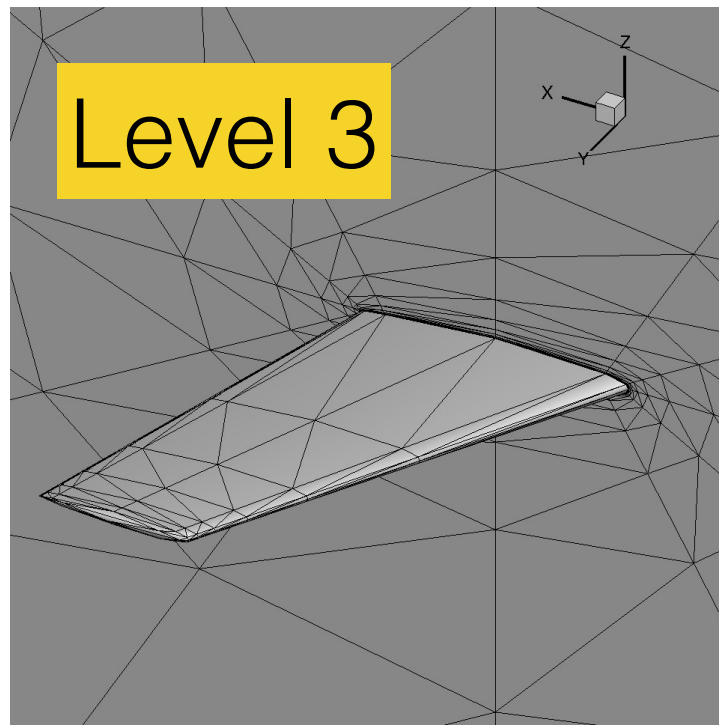
Finest=Level1



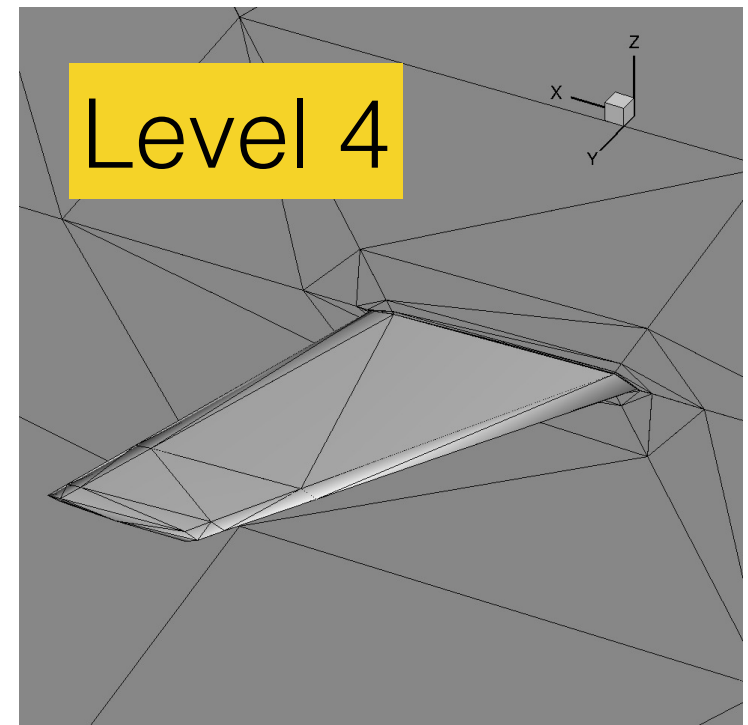
Level 2



Level 3



Level 4



Inter-Grid Nodal Interpolation

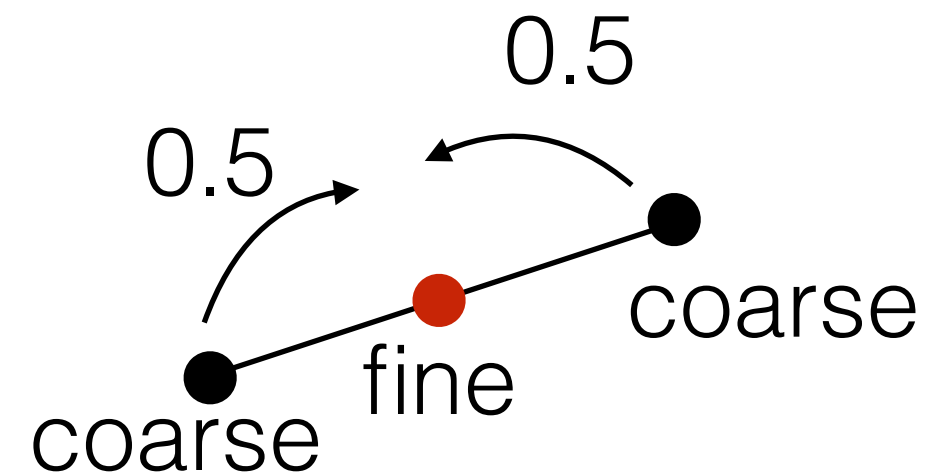


Inter-grid nodal interpolation information is given in two files:
.prolong_nc_seq (nested nodes) .prolong_nc (others)

Pure tetrahedral grids:

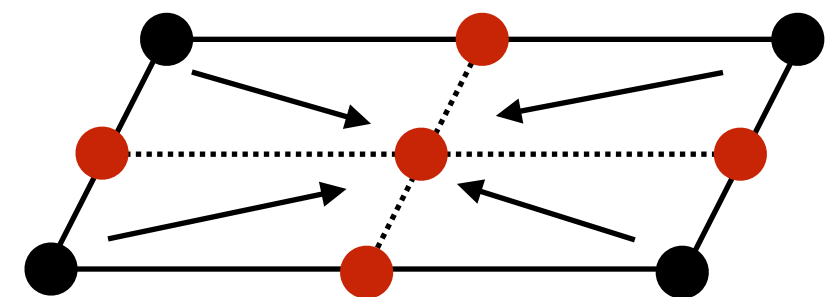
Injection at nested nodes.
Average over coarse-nghbr nodes.

Note: Find-grid node is not necessarily at the midpoint
of the coarse-grid edge.



Other grids:

Injection at nested nodes.
Average over coarse-nghbr nodes.
Average over nearest coarse-grid nodes (if no nghbrs).

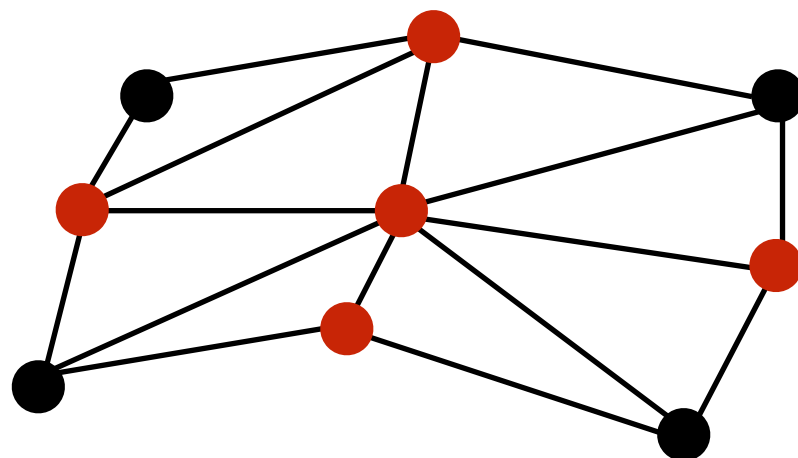


‘Midpoint’ data

.midp file is generated for each coarser grid (tetra only).

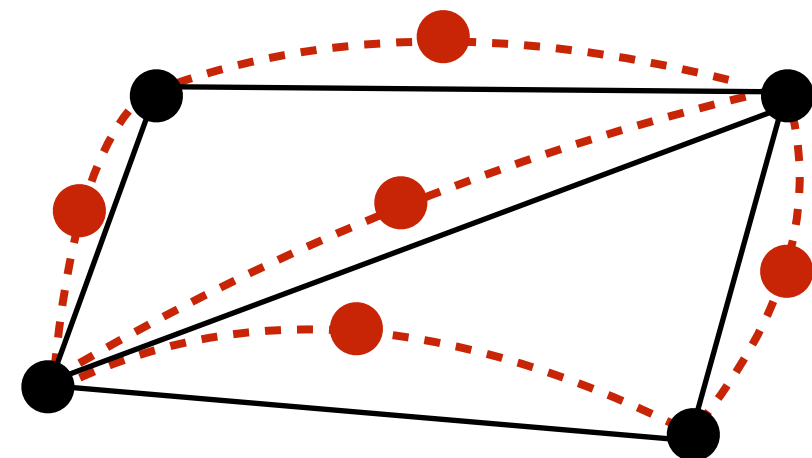
This file contains a list of nodes removed from the finer grid, which correspond to nodes between coarse-grid edges (not exactly the edge-midpoint). These nodes are on the true geometry on the boundary. So, P2 tetrahedral mesh can be constructed for all coarser grids.

P1 fine grid



coarsening
→

P1 coarse grid (black) + .midp (red)



P2 grid indicated in red.